



Effect of the plate bondstress-slip property on the flexural strength of FRP Plated RC beams using a displacement-based approach

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Abstract

This study presents the theoretical results of a displacement-based approach for CFRP plated RC beams while varying the bondstress-slip relationship of the plate to see the influence of this material property on the global behavior. The experimental programme included three ordinary and three plated RC beams that had similar dimensions and similar material properties. From the displacement-based analysis, it was shown that the theoretical and experimental results for the ordinary RC beams were almost identical. However, for the CFRP plated RC beams, it was shown that using a linear-descending bondstress-slip model for the FRP plate in the analysis underestimated the flexural capacity of the plated beam as the theoretical results were about 75% of the experimental values. Moreover, through a parametric study, it was shown that adding an ascending branch to the FRP plate's bondstress-slip property (e.g., using a bilinear bondstress-slip model) results in better correlation with the experimental values at service loads in comparison to using only a linear-descending branch without an ascending component. In addition, it was also shown that adding a frictional component to the bondstress-slip model of the FRP plate to allow the force in the plate to build up as debonding ensues gives closer results to the experimental values observed in the laboratory. The accurate simulation of the FRP plate's bondstress-slip material property was seen to greatly affect the plated member's capacity and ductility behavior.

Keywords Adhesively plated beam · FRP · Intermediate crack debonding · Displacement-based analysis · Partial-interaction · Bondstress-slip

1 Introduction

Enhancing the flexural capacity and service life of RC members through adhesively bonded fibre-reinforced polymer (FRP) plates/sheets has become a common practice recently [1, 2]. The analysis of such structures is typically carried out using a strain-based moment-curvature approach [3, 4], known as full-interaction, by which detachment of the plate can occur due to: (1) intermediate crack (IC) debonding [5–7]; (2) critical diagonal crack (CDC)

debonding [8]; or (3) plate end (PE) debonding [9]. It is typical to design FRP plated RC structures for IC debonding first and then check for CDC and PE debonding [10]. As such, the full-interaction analysis is usually halted when the IC debonding strain is first observed in the plate. Following such design approach, debonding of the plate often occurs before the plate can achieve its full tensile strength and sometimes before the yielding of the steel rebars [11].

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This has inspired several researchers to look into partial-interaction theory to predict the flexural strength of FRP plated members instead of the basic full-interaction methods. In essence, partial-interaction refers to the slip or sliding that occurs between steel bars or plate and surrounding concrete when a crack transverses the reinforcement. Furthermore, partial-interaction is not only restricted to sliding between the reinforcement and concrete, but can account for sliding between two concrete surfaces as well, and this is known as shear friction [12]. To quantify the force in the reinforcement due to sliding against the adjacent concrete, researchers carried out tension stiffening analysis of concrete prisms, and both numerical methods [13] and closed-form solutions [14] were developed. In addition to that, shear friction theory was utilized to determine the compressive forces in concrete while accounting for both the elastic and inelastic deformations caused by sliding of the concrete wedge [15, 16]. To link the tensile and compression regions of the RC structure, a moment-rotation analysis at an RC hinge was first performed [17]; and later moment-rotation analysis of RC beam segments were presented as well which accounted for the formation of multiple cracks [18].

Quantifying the moment-rotation relationship of RC segments allowed researchers to come up with mechanics-based solutions to calculate the mid-span deflection of RC beams [19, 20], non-prismatic RC beams [21] and FRP plated RC beams [22]. The research on partial-interaction design of RC structures progressed such that displacement-based mechanics solutions for predicting the flexural capacity of FRP plated RC beams were presented [11, 23, 24]. The displacement-based approach considered the plate’s IC debonding mechanism and this

is illustrated by the FRP plated prism depicted in Fig. 1. In such case, failure does not occur when the loaded end of the plate achieves the IC debonding strain as shown in Fig. 1a; instead, the plate continues to extend, allowing for the propagation of debonding in the plate as shown in Fig. 1b. Failure through IC debonding ultimately occurs when the entire plate undergoes debonding (Fig. 1c). Similarly, when the moment in the FRP plated beam in Fig. 2 reaches M_{IC} , the force in the plate becomes P_{IC} which induces the initiation of IC debonding as in Fig. 1a. In case the moment is increased further, say to M_1 in Fig. 2a, part of the plate debonds as in Fig. 1b and the elongation within the debonded region becomes $\delta_{max} + \epsilon_{IC} L_{db-p}$. Within this debonded region, the plate is not bonded to the concrete surface and acts as a passively prestressed tendon exerting a force P_{IC} on the RC member at the position of the plate [11]. This theoretical novel approach, presented by Oehlers et al. [11], allowed for the plated beam to carry greater loads, but was not verified against experimental results.

In addition, numerous researchers have investigated the bondstress-slip relationship for externally bonded plates [25–27] and many more have tried to measure the FRP-to-concrete interface bond strength [28–30]. In doing so, researchers have quantified the FRP-to-concrete bond characteristics using double-shear or single-shear pull-push tests, or beam tests [29]. In all of these tests, the concrete prism length is usually marginally larger than the plate’s effective length (or critical length) and typical bondstress-slip relationships are depicted in Fig. 3. Given that the slip at maximum bondstress, δ_1 , for FRP plated beams is much smaller than the maximum debonding slip, δ_{max} , it is common to simplify the analysis by adopting

Fig. 1 IC debonding mechanism [11]

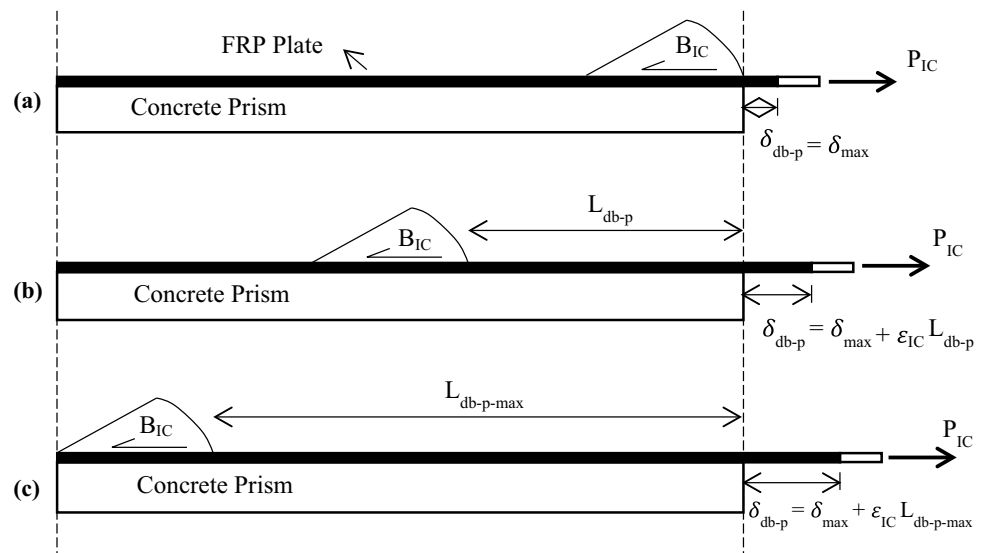


Fig. 2 Debonding mechanism of a symmetrically loaded FRP plated member [11]

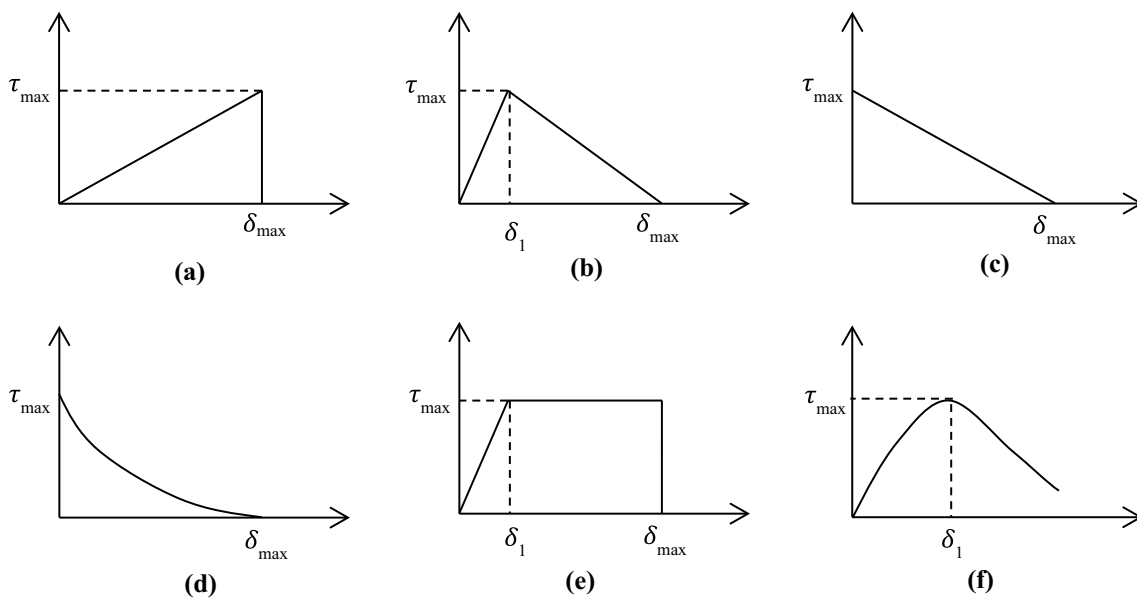
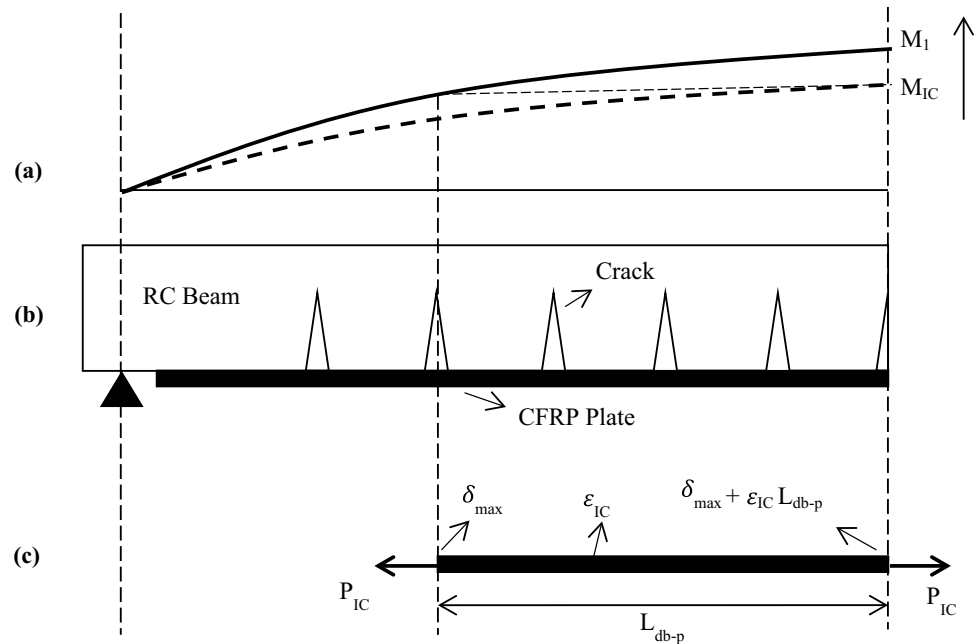


Fig. 3 Different bondstress-slip models

a linear-descending relationship as shown in Fig. 3c [31], especially when the ultimate capacity of the RC beam is required.

The bondstress-slip models proposed by previous researchers hardly have a residual frictional component; however, Zhang and Smith [32] state that a residual friction (or shear) component, f_r , as shown in Fig. 4, may develop in the FRP-to-concrete interface when either: (a) the bonded length of the plate in the concrete prism is more than the plate’s critical length; (b) the plate length in

flexural members is more than the plate’s critical length; or (c) anchorage devices are installed in the plate. Henceforth, for FRP plated RC beams, a frictional component may indeed form in the plate as it undergoes deformation due to loading as most plated RC members have lengths in excess of the critical length of the plate. Therefore, the objectives of this publication are: (a) to compare the results of the theoretical displacement-based approach with experimental results; (b) to determine the effect of the bondstress-slip relationship on the global behavior of the FRP plated beam

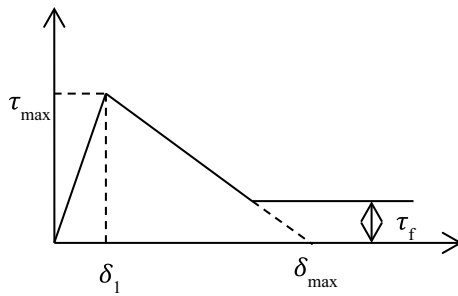


Fig. 4 Bondstress-slip model with frictional component [32]

using a displacement-based approach, and to see whether a frictional component develops in the FRP plate as it undergoes deformation prior to complete debonding. As such, the related segmental analysis and displacement-based analysis of the FRP plated member are presented first. After that, the experimental programme is given which includes three ordinary RC beams and three CFRP plated RC beams. Next, the results of the displacement-based analysis are presented and compared with the experimental values. Finally, a parametric study is given to show the effect of altering the bondstress-slip property on the global behavior of the FRP plated member (e.g., adding a frictional component or adding an ascending branch).

2 Segmental analysis

In the ensuing segmental analysis, the concrete full stress-strain relationship was determined using Popovics [33] expression as follows;

$$\frac{\sigma_{ax}}{f_{cc}} = \frac{\left[\frac{(\epsilon_{ax})_{200}}{\epsilon_{cc}} \right]^r}{r - 1 + \left[\frac{(\epsilon_{ax})_{200}}{\epsilon_{cc}} \right]^r} \tag{1}$$

where σ_{ax} is the axial stress, $(\epsilon_{ax})_{200}$ is the 200 mm cylinder axial strain, f_{cc} is the confined concrete peak stress, ϵ_{cc} is the confined concrete strain at peak stress, and r is a ductility factor. Both f_{cc} and ϵ_{cc} , together with r in Eq. (1) can be determined as follows [16];

$$f_{cc} = f_{co} \left[1 + 3.65 \left(\frac{\sigma_{con}}{f_{co}} \right)^{0.82} \right] \tag{2}$$

$$\epsilon_{cc} = \epsilon_{co} \left[1 + 14.74 \left(\frac{\sigma_{con}}{f_{co}} \right)^{1.03} \right] \tag{3}$$

$$r = \frac{E_c}{E_c - \frac{f_{cc}}{\epsilon_{cc}}} \tag{4}$$

where f_{co} is the unconfined concrete peak stress, ϵ_{co} is the unconfined concrete strain at peak stress, σ_{con} is the confinement stress, usually provided by the stirrups, and E_c is the concrete elastic modulus. Chen et al. [15] give ϵ_{co} as;

$$\epsilon_{co} = 4.76 \times 10^{-6} f_{co} + 2.13 \times 10^{-3} \tag{5}$$

The confinement stress, σ_{con} , can be calculated using the following expression [17];

$$\sigma_{con} = \frac{n(f_{y-link})(A_{link})}{b(S_{link})} \tag{6}$$

where f_{y-link} is the steel stirrups' yield stress, A_{link} is the cross-sectional area of the steel stirrups, n is the number of legs in the steel stirrups, S_{link} is the spacing between different stirrups, and b is the width of the beam. In addition, the ACI code [34] give E_c as;

$$E_c = 3320 \sqrt{f_{co}} + 6900 \tag{7}$$

where f_{co} in Eqs. (5) and (7) is in MPa. The size and shape effect of the concrete segment in compression, as well as the formation of the concrete wedge, were accounted for following Chen et al. [15] and Chen et al. [16] method. For instance, the size-dependent axial global strain, ϵ_{axgl} , was taken as;

$$\epsilon_{axgl} = \left[(\epsilon_{ax})_{200} - \epsilon_{mat} \right] \frac{100}{L_{def}} + \epsilon_{mat} \tag{8}$$

where ϵ_{mat} is the elastic material strain and L_{def} is half the segment length.

The reinforcement steel stress-strain relationship was idealized by the model illustrated in Fig. 5a [20]. Such model accounts for the elastic response of the reinforcement and the post yield response as well. Moreover, the stress-strain response depicted in Fig. 5b was used to represent the constitutive relationship for the CFRP plate [3, 22].

Furthermore, the bondstress-slip model presented by Haskett et al. [13] and depicted in Fig. 6a was adopted for the reinforcement steel to represent local deformations. As for the CFRP plate, the linear-descending bondstress-slip model presented by Seracino et al. [31] and shown in Fig. 6b was used in the analysis. Generic expressions for the maximum slip, δ_{max} , and maximum bondstress, τ_{max} , for a given plate configuration are given as follows [31];

$$\delta_{max} = \frac{0.976 \varphi_f^{0.526}}{0.802 + 0.078 \varphi_f} \tag{9}$$

$$\tau_{max} = (0.802 + 0.078 \varphi_f) f_c^{0.6} \tag{10}$$

where

$$\varphi_f = \frac{t_d}{2 t_b + b_p} \tag{11}$$

and t_d and t_b in Eq. (11) can be taken as 1 mm [31], and b_p is the width of the plate.

2.1 Uncracked segmental analysis

Initially, an uncracked segmental analysis takes place as shown in Fig. 7 which is basically a full-interaction analysis. Half the segment is rotated by an angle θ as in Fig. 7, and the depth of the neutral axis, d_{NA} , is first assumed. Next, the segment's curvature, χ , can be determined as follows;

$$\chi = \frac{\theta}{L_{def}} \tag{12}$$

where L_{def} is half the segment length as illustrated in Fig. 7, and angle θ is in radians. Having determined the curvature of the segment, the strain in the furthest compression and tensile fibres can be calculated accordingly as follows;

Fig. 5 Idealized stress-strain relationship: (a) steel rebars; (b) FRP plate

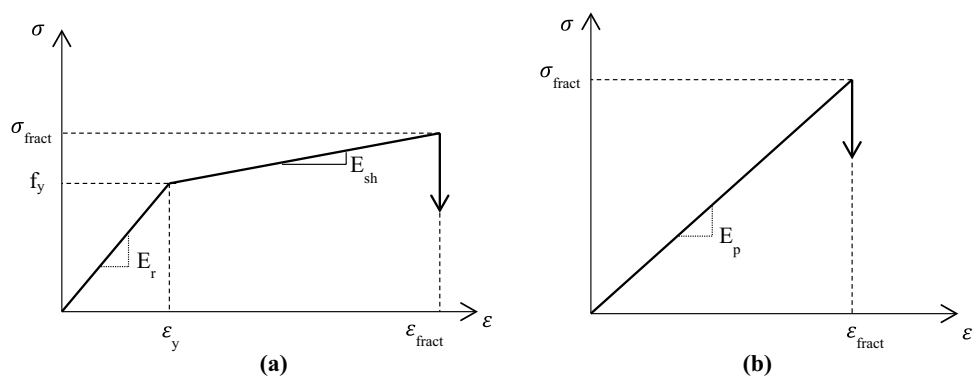


Fig. 6 Bondstress-slip characteristic: (a) steel rebars; (b) FRP plate

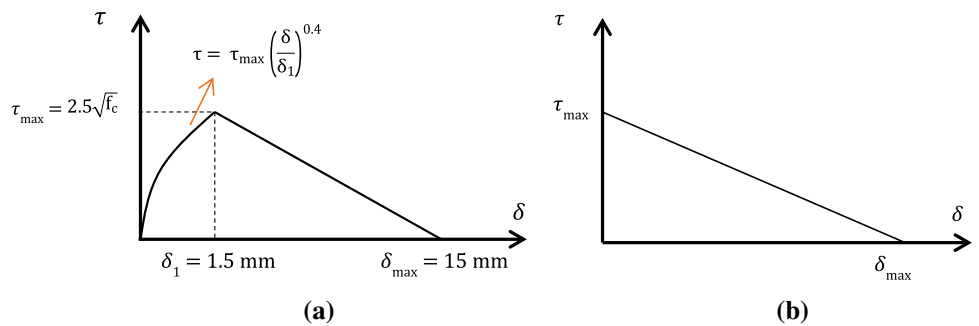
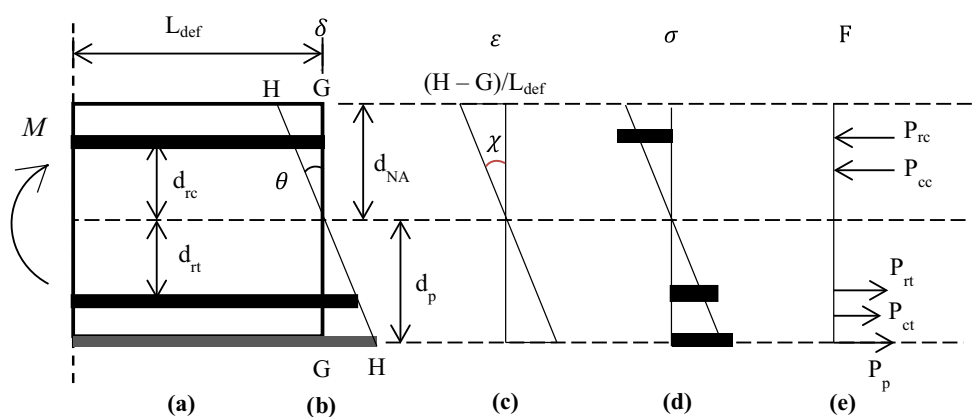


Fig. 7 Uncracked segmental analysis



$$\epsilon_{c-c} = \chi d_{NA} \tag{13}$$

$$\epsilon_{c-t} = \chi (h - d_{NA}) \tag{14}$$

Similarly, the strain in the compression rebars, tension rebars, and FRP plate can be determined as follows;

$$\epsilon_{rc} = \chi d_{rc} \tag{15}$$

$$\epsilon_{rt} = \chi d_{rt} \tag{16}$$

$$\epsilon_p = \chi d_p \tag{17}$$

where d_{rc} , d_{rt} , and d_p are as shown in Fig. 7.

The stress distribution in the uncracked segment is illustrated in Fig. 7d. A triangular stress distribution initially governs the concrete in compression and tension as shown. From the concrete stress diagram, the concrete compressive force, P_{cc} , and concrete tensile force, P_{ct} can be derived accordingly. P_{cc} acts at $1/3 d_{NA}$ from the top of the segment while P_{ct} acts at $1/3 (h - d_{NA})$ from the bottom of the segment. Moreover, as the strains in the reinforcement bars and FRP plate have already been determined, the reinforcement compressive force, P_{rc} , the reinforcement tensile force, P_{rt} , and the FRP plate force, P_p can be determined from their material properties. Having determined all the forces acting along the segment, the goal is to find the depth d_{NA} that would ensure force equilibrium for a given rotation, after that, the moment of the segment, M , can be calculated. Having done so, the rotation, is changed and the new moment M is determined. The above analysis is continued until a tensile crack forms in the segment.

2.2 Single crack segmental analysis

At this point, the analysis starts with a single crack segmental analysis shown by the initial crack in Fig. 8. The closed-form solution for the force P_{rt} following a single-crack segmental analysis is given as follows [14];

$$P_{rt} = A_{rt} E_{rt} \sqrt{\frac{2 \lambda_2 \Delta_{rt}^{1+\phi}}{1 + \phi}} \tag{18}$$

where

$$\lambda_2 = \frac{\beta_2 \tau_{max}}{\delta_1^\phi} \tag{19}$$

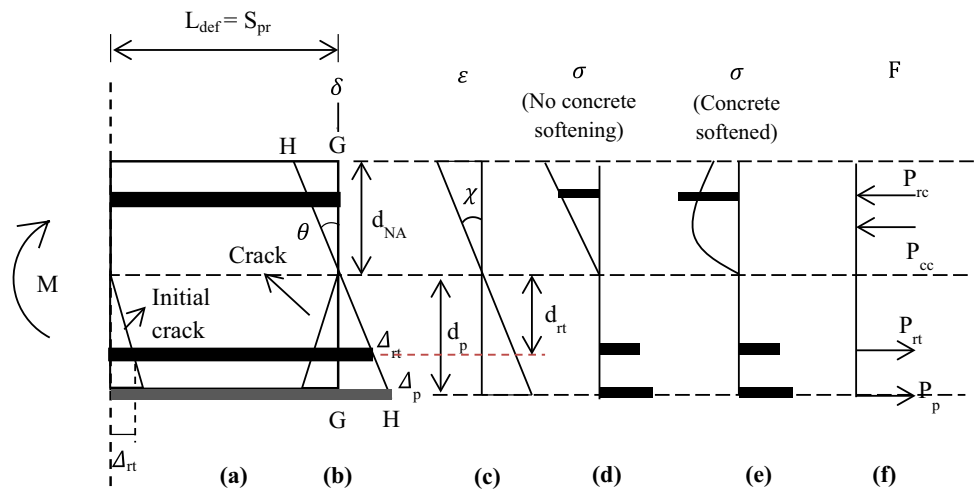
$$\beta_2 = \frac{L_p}{A_{rt}} \left(\frac{1}{E_{rt}} + \frac{A_{rt}}{E_c A_c} \right) \tag{20}$$

E_{rt} and A_{rt} in Eq. (18) are the Young's modulus and the cross-sectional area of the tensile reinforcement, ϕ is a factor equal to 0.4, L_p is the sum of the perimeter of the tensile reinforcement, and A_c is the area of the concrete prism taken as $(b \times 2c)$ where c is the measurement from the middle of the tensile reinforcement to the bottom tensile fibre. Equation (18) applies while the reinforcement is still elastic; however, when the reinforcement yields, the closed-form solution is given as follows [20];

$$P_{rt} = A_{rt} E_{sh} \sqrt{\frac{2 \lambda_{sh} (\Delta_{rt} - \Delta_y)^{1+\phi}}{1 + \phi}} + A_{rt} f_y \tag{21}$$

where

Fig. 8 IC debonding segmental analysis



$$\lambda_{sh} = \frac{\beta_{sh} \tau_{max}}{\delta_1^\varphi} \tag{22}$$

$$\beta_{sh} = \frac{L_p}{A_{rt}} \left(\frac{1}{E_{sh}} + \frac{A_{rt}}{E_c A_c} \right) \tag{23}$$

f_y is the steel yielding stress, E_{sh} is the strain-hardening modulus as shown in Fig. 5a and β_y in Eq. (21) is given by;

$$\frac{\lambda_2 \Delta_y^{1+\varphi}}{1+\varphi} = 0.5 \left(\frac{f_y}{E_{rt}} \right)^2 \tag{24}$$

As for the force in the FRP plate, following a linear-descending bondstress-slip model as in Fig. 6b, the closed-form solution is given by [14];

$$P_p = A_p E_p \lambda_3 \delta_{max} \sin \left[\arccos \left[\frac{\delta_{max} - \Delta_p}{\delta_{max}} \right] \right] \tag{25}$$

where

$$\lambda_3 = \sqrt{\frac{\tau_{max}}{\delta_{max}}} \beta_2 \tag{26}$$

and E_p and A_p are the modulus of elasticity and cross-sectional area of the FRP plate. The concrete compressive force, P_{cc} , in Fig. 8f can be determined by taking the area under the stress diagram in Fig. 8e. Equation (1) can be used to plot the concrete stress diagram while considering the softening branch as well, while Eq. (8) can be used to account for the size of the concrete segment. As before, force equilibrium is first sought after by fixing the rotation and changing d_{NA} until the summation of forces is equal zero. After that, the moment of the segment is determined. The same procedure is repeated again with a different segment rotation and the resulting moment is subsequently recorded.

2.3 Multiple crack segmental analysis

As the tensile reinforcements' force builds up, further flexural cracks form at distance S_{pr} from the original crack, and the behavior reverts to that of a multiple crack segmental analysis as illustrated in Fig. 8. The distance S_{pr} can be calculated as follows [14];

$$S_{pr} = \left[\frac{(1+\varphi) f_{ct} A_c \delta_1^\varphi}{\tau_{max} L_p (\sqrt{2C_9})^\varphi} \right]^{\frac{1}{1+\varphi}} \tag{27}$$

where

$$C_9 - \frac{\sqrt{2C_9} f_{ct} A_c \lambda_2 \delta_1^\varphi}{\tau_{max} L_p} = -0.5 \left(\frac{f_{ct}}{E_c} + \frac{f_{ct} A_c}{A_{rt} E_{rt}} \right)^2 \tag{28}$$

and f_{ct} is the concrete tensile strength. The closed-form solution for the tensile reinforcement force, P_{rt} , following the CEB-FIP bondstress-slip ascending branch (Fig. 5a), when multiple cracks form prior to and after yielding of the reinforcement are given as follows [20];

$$P_{rt} = A_{rt} E_{rt} \sqrt{\frac{2 \lambda_2 \Delta_{rt}^{1+\varphi}}{1+\varphi} + \left(\frac{\Delta_{rt}}{S_s} \right)^2} \tag{29}$$

$$P_{rt} = A_{rt} E_{sh} \sqrt{\frac{2 \lambda_y (\Delta_{rt} - \Delta_{y-p})^{1+\varphi}}{1+\varphi} + \left(\frac{\Delta_{rt} - \Delta_{y-p}}{S_s} \right)^2} + A_{rt} f_y \tag{30}$$

where Δ_{y-p} can be determined from the following expression,

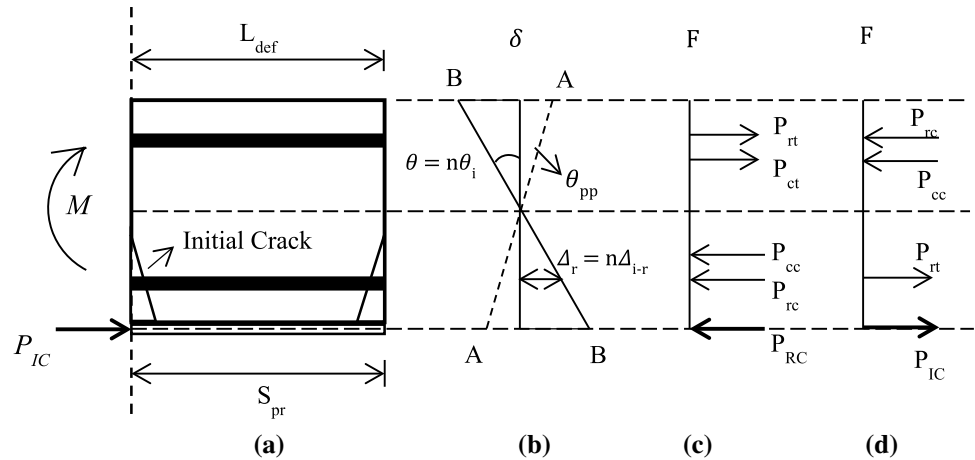
$$\frac{\lambda_2 \Delta_{y-p}^{1+\varphi}}{1+\varphi} + 0.5 \left(\frac{\Delta_{y-p}}{S_s} \right)^2 = 0.5 \left(\frac{f_y}{E_{rt}} \right)^2 \tag{31}$$

and S_s is equal to $S_{pr}/2$. As for the FRP plate, a single-crack analysis dominates since the behavior of EB FRP plates ultimately revert back to single crack analysis. The closed-form solution for the FRP plate is given by Eq. (25); however, the slip of the plate is taken as the entire slip across L_{def} which is 2_p in Fig. 8a. As for the concrete in compression, P_{cc} , is determined as before while considering the size of the segment and softening of the concrete in compression. For each rotation, the resulting moment is determined, and the analysis is continued until a rotation causes the slip of the plate, p_r to reach p_{max} . The corresponding moment when p_r equals p_{max} is the moment that causes the plate's IC debonding and is commonly known to as the IC debonding moment, M_{IC} . After this moment in the beam is achieved, the plate is assumed to be detached from the concrete surface [11], and the analysis is continued following a passively prestressed segmental analysis as depicted in Fig. 9.

2.4 Passively prestressed segmental analysis

The passively prestressed segment is subjected to P_{IC} at plate's level. In case this force is applied first prior to any deformation, this will cause the displacement A-A shown in Fig. 9b with rotation θ_{pp} . Henceforth, it becomes a question of determining the resultant force P_{RC} ($P_{rc} + P_{cc} - P_{rt} - P_{ct}$) at the plate's level that is equivalent to P_{IC} and a zero moment at the same level. In case the concrete in tension cracks, then P_{rt} in Fig. 9c will have to be determined using partial-interaction.

Fig. 9 Passively prestressed segmental analysis



The passively prestressed segmental analysis is continued by applying a moment M that causes the displacement of the segment to change to that of line B-B in Fig. 9b. At this stage, a single crack shown as the initial crack in Fig. 9a takes place. The force profile is now given by that in Fig. 9d and the resultant of the forces should be equivalent to P_{IC} ; after which, the segment's moment can be measured at the plate's level. Once P_{rt} reaches the force to cause primary cracks, P_{pr} , another crack forms at S_{pr} from the former crack as depicted in Fig. 9a, and the force in the steel rebars is now governed by a multiple crack analysis. Again, the section is rotated until the resultant force, P_{RC} , is equivalent to the IC debonding force, P_{IC} , after which the moment at the level of the plate can be calculated.

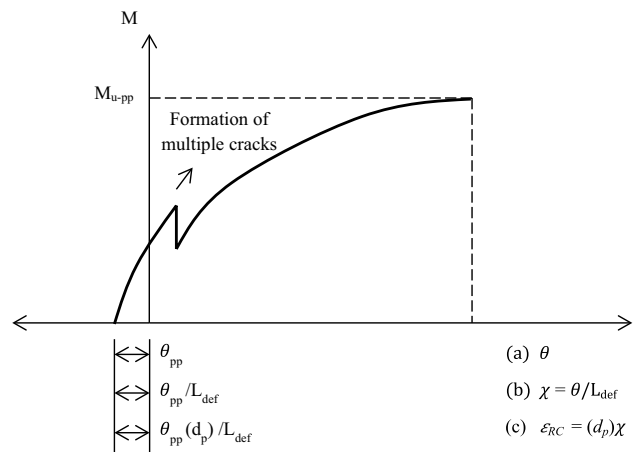


Fig. 10 Moment relationship of a passively prestressed segment: (a); (b); (c)

3 Displacement-based flexural capacity of FRP plated RC beams

The moment-rotation (M -) analysis described above can be utilized to plot the M - relationship of a passively prestressed segment as shown in Fig. 10a. Using the M - plot, the moment-curvature (M -) relationship can be calculated by dividing the rotation by the concrete segment's half-length, L_{def} as illustrated in Fig. 10b. Also, the concrete strain at plate's level, ϵ_{RC} (the product of the segment's curvature and the plate's depth from the neutral axis, d_p) can be plotted against the moment as shown in Fig. 10c. As the M - in the Fig. 10 is derived from a passively prestressed segment as in Fig. 9, a passively prestressed rotation θ_{pp} exists as shown in in Figs. 9b and 10 when $M=0$. This arises due to the passively prestressed force P_{IC} , which requires rotation θ_{pp} for the segment to be in equilibrium.

Figure 11c shows the plate displacement over the half-length of a symmetrically loaded adhesively plated beam. This displacement of the plate can be calculated as [11];

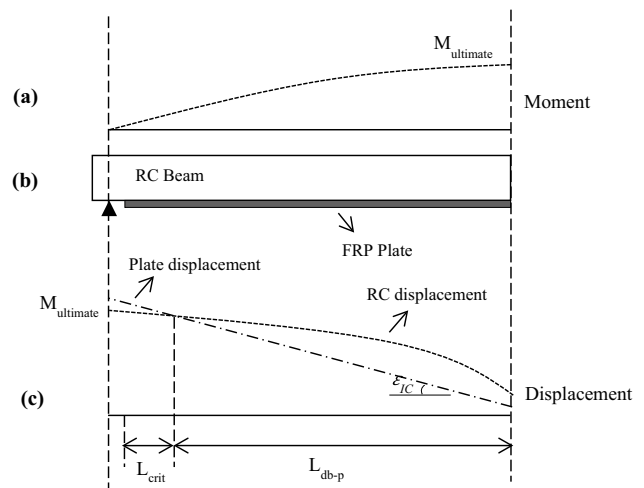


Fig. 11 FRP plated beam flexural capacity

$$\delta_{db-p} = \delta_{max} + \epsilon_{IC} L_{db-p} \tag{32}$$

where ϵ_{IC} is the IC debonding strain, and L_{db-p} is the debonded length of the plate. In addition, Fig. 11c illustrates the concrete displacement at the plate’s level for a given applied moment, M , on the beam. The concrete displacement can be given by [11];

$$\delta_{db-RC} = \int_0^{L_{db}} \epsilon_{RC} \tag{33}$$

where ϵ_{RC} can be determined from Fig. 10c. The ultimate moment ($M_{ultimate}$) of the adhesively plated beams is achieved when the plate’s displacement, given by Eq. (32), equals the concrete displacement at the plate’s level, given by Eq. (33), at distance L_{crit} from the plate end, satisfying the conditions of displacement compatibility at the plate’s level (Fig. 11c). The critical length of the plate, L_{crit} , can be calculated as follows [31];

$$L_{crit} = \frac{\pi}{2\lambda} \tag{34}$$

where

$$\lambda^2 = \frac{\tau_{max} L_p}{\delta_{max} E_p A_p} \tag{35}$$

4 Experimental programme

A total of six RC beam specimens were considered here, three of those were carbon fibre-reinforced polymer (CFRP) plated RC beams tested by El-Zeadani et al. [22].

The remaining three beam specimens, tested by El-Zeadani et al. [20], were identical to the plated RC beams but without the CFRP plates to ensure the validity of the mechanics-based solution in the absence of the FRP plate. A brief description of the beam specimens, materials and test setup is given below.

4.1 Beam details

The adhesively plated beams (CFRP-B1, CFRP-B2 and CFRP-B3), illustrated in Fig. 12, had a total span of 3600 mm and a 3500 mm span between the supports. The tension face CFRP plate had a total length of 3250 mm, a thickness of 1.2 mm and a width of 50 mm. Furthermore, the beams had 16 mm and 12 mm diameter tensile and compression steel reinforcement, respectively, as illustrated in Fig. 12a. Also, the plated beams had 8 mm diameter mild steel shear links spaced at 150 mm. As for the unplated RC beams (RC-B1, RC-B2 and RC-B3), they had similar properties to the adhesively plated ones except for the fact that no CFRP plate was attached to their tension face and their details are shown in Fig. 13.

4.2 Materials

All six beam specimens were casted on the same day and had similar material properties. The 150 × 150 mm 28th day mean concrete cube compressive strength was 26.3 MPa. The 150 × 300 mm diameter and the 100 × 200 mm diameter mean cylinder compressive strengths were 20.5 MPa and 22.7 MPa, respectively. Furthermore, Table 1 gives the material properties of the reinforcement.

Fig. 12 Adhesively plated beam specimens’ details

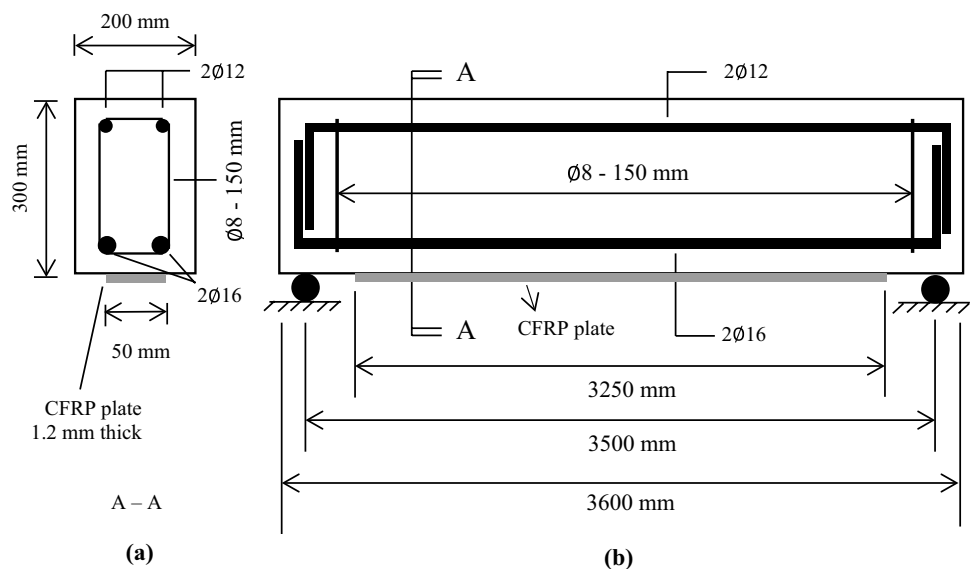


Fig. 13 Unplated beam specimens' details

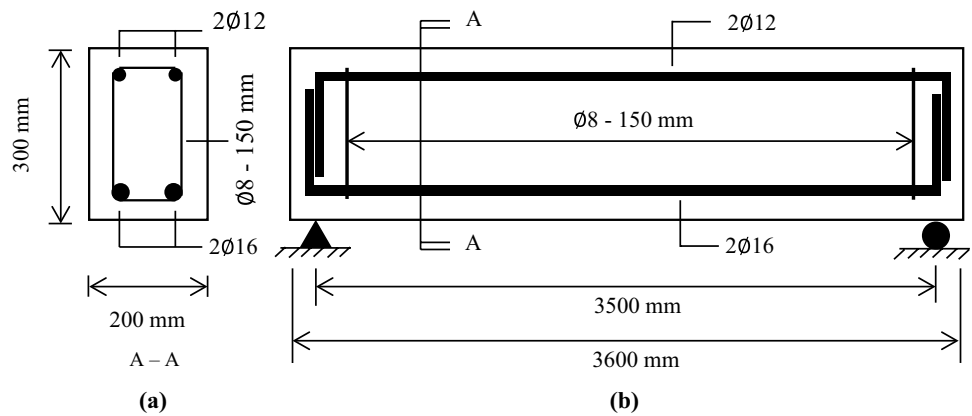


Table 1 Reinforcement mechanical properties

Property	Reinforcement steel	CFRP plate steel
Elastic modulus (GPa)	154	165
Strain-hardening modulus (GPa)	1.57	–
Yield stress, f_y (MPa)	535	–
Fracture tensile strength (MPa)	615	2800

4.3 Experimental setup

Using a hydraulic jack, a concentrated load at mid-span was applied on the beams until failure. After each increment, the mid-span strains in the steel rebars and CFRP plate were measured as the test proceeded using strain gauges. The experimental setup is illustrated in Fig. 14. As for the unplated beams, a similar test setup was used in the absence of the plate.

5 Results and discussion

In the following subsections, the experimental and theoretical RC unplated and plated beam results are presented and a comparison between the two is given. This includes the mid-span strain in the steel reinforcement, mid-span strain in the EB plate and ultimate loads.

5.1 Unplated RC beams

The mechanics-based solution for the unplated beams was similar to what has been described above except for the fact that no CFRP plate was considered in the analysis. The theoretical and experimental mid-span load-strain results for the steel rebars are presented in Fig. 15 [20]. Clearly, a significant correlation between the two results exists with yielding occurring at a load of about 55 kN (except that for RC-B2). Moreover, Fig. 16 shows the M- relationship for an

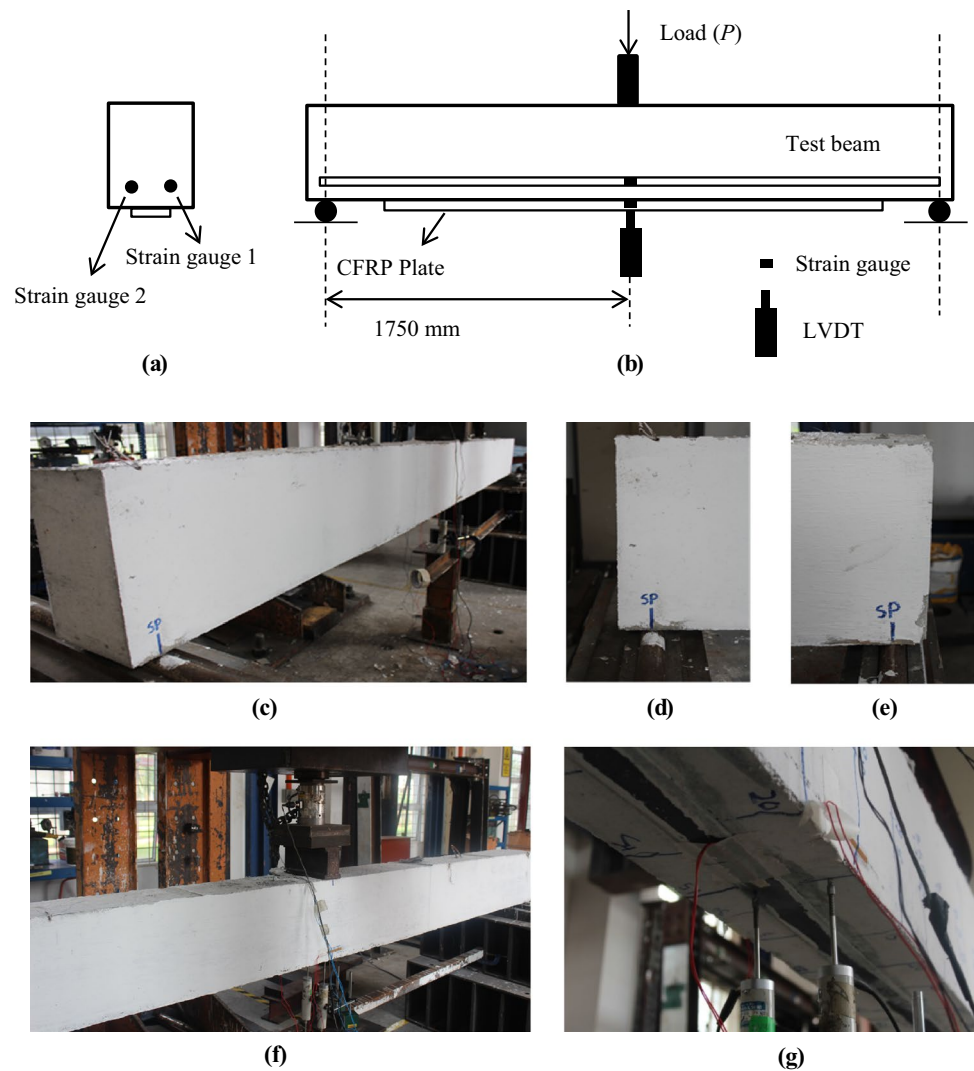
RC beam segment extracted from the unplated RC beam shown in Fig. 13. The RC segment was 290.6 mm long and the moment-rotation analysis was halted when the reinforcement reached its fracture strain, f_{fract} at $M = 57.1$ kN-m. Accounting for the self-weight of the beam (concrete unit weight, $c = 25$ kN/m³), the theoretical ultimate load of the unplated RC beams was 62.6 kN. Table 2 summarizes the experimental and theoretical ultimate loads for the three RC beam specimens to which it can be deduced that the mechanics-based solution provides a good estimate of the ultimate load for unplated RC beams as the difference between the theoretical and the average experimental ultimate load was about 0.6%.

5.2 Adhesively plated RC beams

The theoretical mid-span applied load versus strain response for the steel reinforcement bars were contrasted with those observed in the laboratory for CFRP-B1, CFRP-B2 and CFRP-B3, and the results are presented in Fig. 17a, b, c, respectively. In all of the three beams, it can be deduced that the elastic slope of the mid-span load-strain response was stiffer for the experimental results as compared to the theoretical ones. Furthermore, from the theoretical load-strain response, the reinforcement steel yielded at a much earlier applied load, P , in contrast to the experimental ones. For instance, the theoretical load-strain response shows that the reinforcement steel yielded at a mid-span load of about 61.55 kN while the experimental results show that yielding occurred at a mid-span load of 68 kN in CFRP-B3 and about 75 kN in CFRP-B2.

As for the mid-span strain in the CFRP plate, the results are given in Fig. 17d. Initially, and prior to cracking of the concrete, the theoretical results were identical to the experimental ones. However, after the formation of flexural cracks, the two results poorly correlate. The slope of the plate's theoretical load-strain response changes three times, first when a single crack intercepts the plate, second when multiple cracks in the concrete segment form and

Fig. 14 Test setup [22]



lastly when the force in the plate reaches P_{IC} after which the force in the plate remains unchanged with increasing displacement. As opposed to the theoretical results, the observed values in the laboratory for the strain in the plate keep increasing until the plate debonds. The slope of the experimental response changes twice, once when cracking of the concrete takes place and another time when the steel reinforcement yields.

As for the theoretical ultimate moment capacity, the M_{-RC} relationship (Fig. 18b) needs to be determined and can be derived from M_{-} relationship for a passively prestressed segment (Fig. 18a) as explained previously. After that, it becomes a question of determining the moment in the beam that will cause compatibility of displacement at the plate's level. L_{crit} was determined to be 115 mm using the expression given by Seracino et al. [31]. Furthermore, the total plate's length was 3250 mm; that is, about 125 mm unplated distance between the support and the free end of the plate exists. Therefore,

compatibility of displacement between the plate and the concrete at the plate's level should be achieved at 240 mm from the support. This is shown in Fig. 19c, where the load, P , at the center of the beam that caused this displacement was 62.5 kN. In other words, the displacement-based approach predicted that debonding failure will occur when the applied load reached 62.5 kN. However, from laboratory tests, the plate debonded at an average load of 83.9 kN (Table 3) and the results obtained from the displacement-based approach heavily underestimate the maximum load prior to failure. The only way to explain this is due to early yielding of the steel reinforcement in the displacement-based analysis, which caused greater deformation at the level of the plate with little increase in load, and also assuming that the force in the plate does not build up after reaching P_{IC} .

From a full-interaction analysis [3] of which I_C is taken as 0.00211, IC debonding occurs at an applied moment of 29.3 kN-m. This in turn results in an ultimate load of

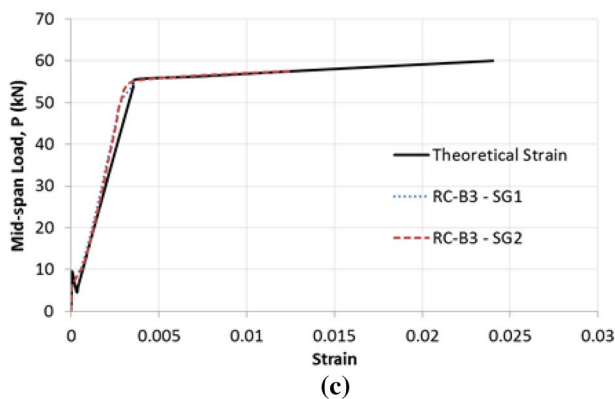
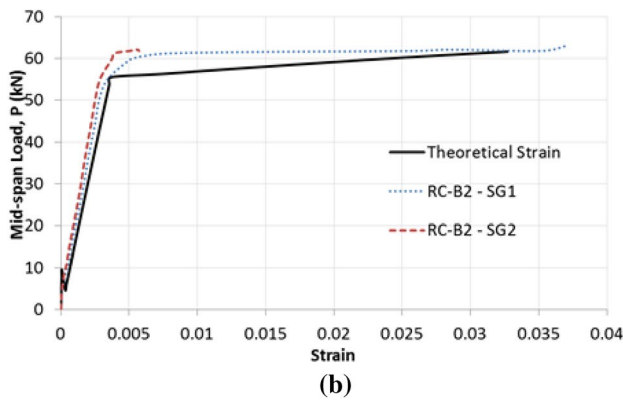
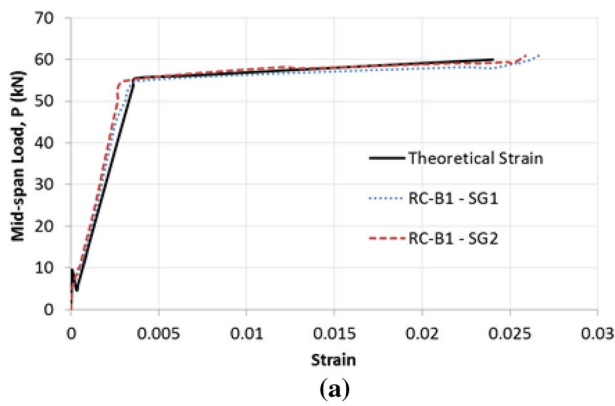


Fig. 15 Reinforcement steel mid-span strain: (a) RC-B1; (b) RC-B2; (c) RC-B3 [20]

30.9 kN while accounting for the self-weight of the beam which is clearly much lower than the actual ultimate load derived from the laboratory, and is less than half of that determined from the displacement-based approach (Fig. 20). Furthermore, for all of the plated beams considered here, failure occurred by progressive IC debonding of the plate, and Fig. 19 depicts the beams at the point of failure.

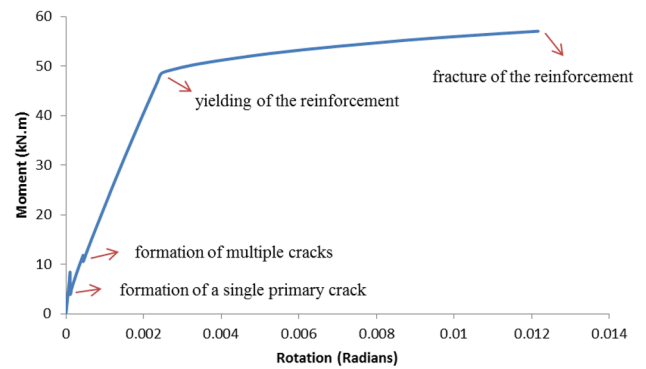


Fig. 16 RC beam M- relationship

6 Parametric study

In the following subsections, a parametric study was conducted to see the effect of adding a frictional component to the plate’s bondstress-slip characteristic on the global behavior of the CFRP plated RC beams (i.e. reinforcement strain, ultimate load). Furthermore, a separate analysis was also conducted to see the effect of adding and altering the ascending branch of the plate’s bondstress-slip model on the behavior of the plated beams.

6.1 Frictional component in the plate’s bondstress-slip model

The main reason why the reinforcement steel in the CFRP plated beams theoretical M- analysis yielded at an earlier applied load to that determined in the laboratory was due to the assumption that the plate’s force remains at P_{IC} once achieved. This in turn caused the reinforcement steel to take up any additional force in the tensile region of the beam due to the segment rotation, while in reality some of that force was shared by the plate. This is indicated by the build-up in the plate strain as shown by the experimental results. Henceforth, for the force in the plate to exceed P_{IC} , the bondstress-slip model (–) needs to allow for shear stresses to develop in the plate when the slip at the crack face exceeds s_{max} .

In a quick parametric study, the plate’s bondstress-slip model was adjusted to allow for a frictional component as shown in Fig. 21. The value for the residual shear strength was chosen to be 1/3 of s_{max} as in Fig. 21a and 1/2 of s_{max} as in Fig. 21b, and the objective was to see their effect on the global behavior of the CFRP plated beams. Since the bondstress-slip models do not close at s_{max} , a closed-form solution for the load-displacement relationship cannot be

Table 2 RC beams experimental and theoretical ultimate loads

Beam specimen	Experimental ultimate load, P_{u-exp} (kN)	Theoretical ultimate load, P_{u-theo} (kN)	P_{u-exp}/P_{u-theo}
RC-B1	62.44	62.6	0.997
RC-B2	64.26	62.6	1.027
RC-B3	62.14	62.6	0.993
Average ultimate load, P_{avg}	62.95	62.6	1.006

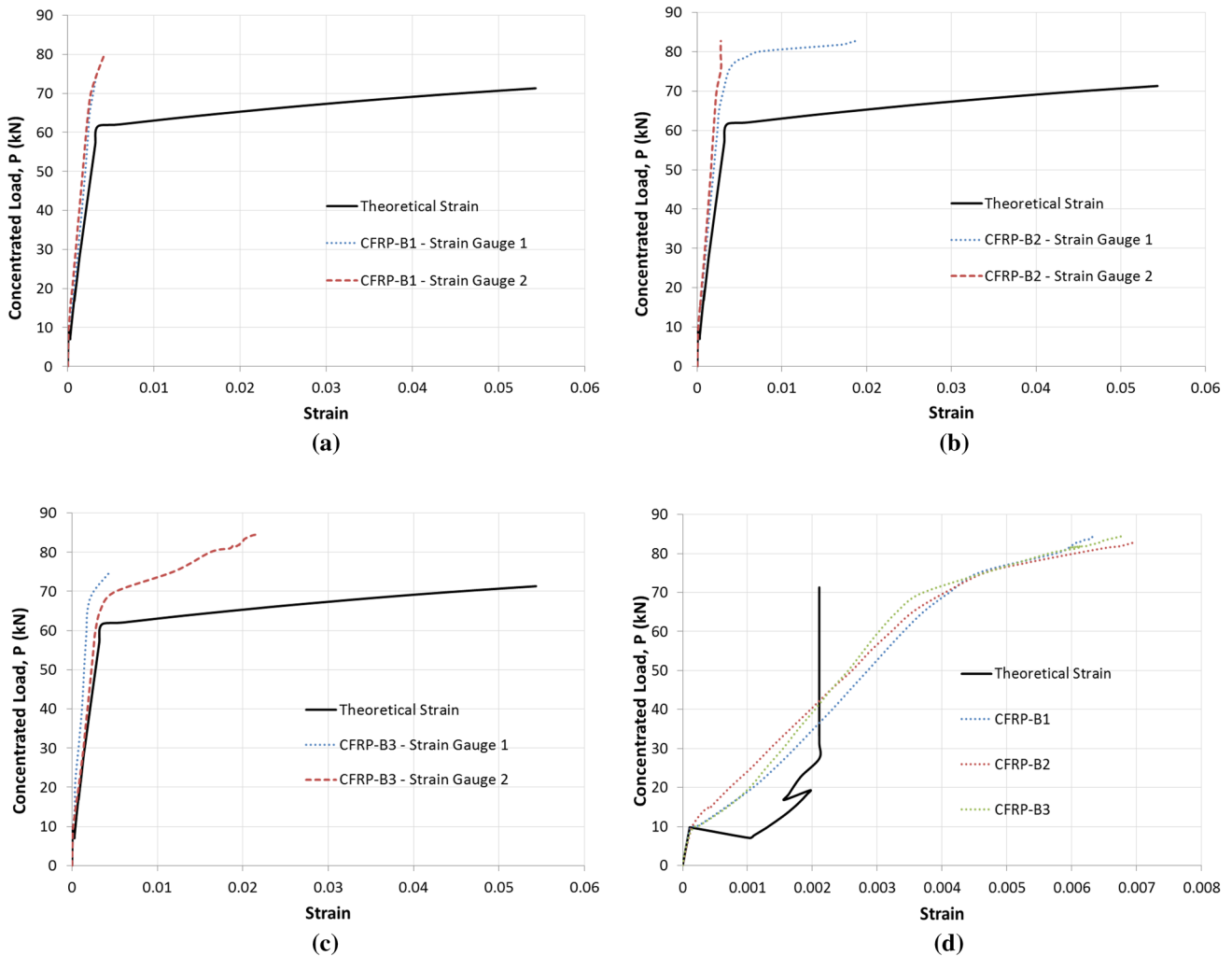


Fig. 17 Mid-span strain results: (a) CFRP-B1 reinforcement steel; (b) CFRP-B2 reinforcement steel; (c) CFRP-B3 reinforcement steel; (d) CFRP plate

developed; instead, Haskett et al. [13] partial-interaction numerical solution which accommodates any shape of the bondstress-slip model can be utilized to determine the load-slip relationship. Other than that, the analysis is similar to what has been described earlier.

The theoretical strains of the reinforcement steel derived from the moment-rotation analysis were

compared with the experimental results of CFRP-B3 and are depicted in Fig. 22a. The line referred to as “Theoretical Strain” was derived from the moment-rotation analysis where the linear-descending bondstress-slip model was used. On the other hand, the lines referred to as “Modified Theoretical Strain 1” and “Modified Theoretical Strain 2” were derived from a moment-rotation analysis in which

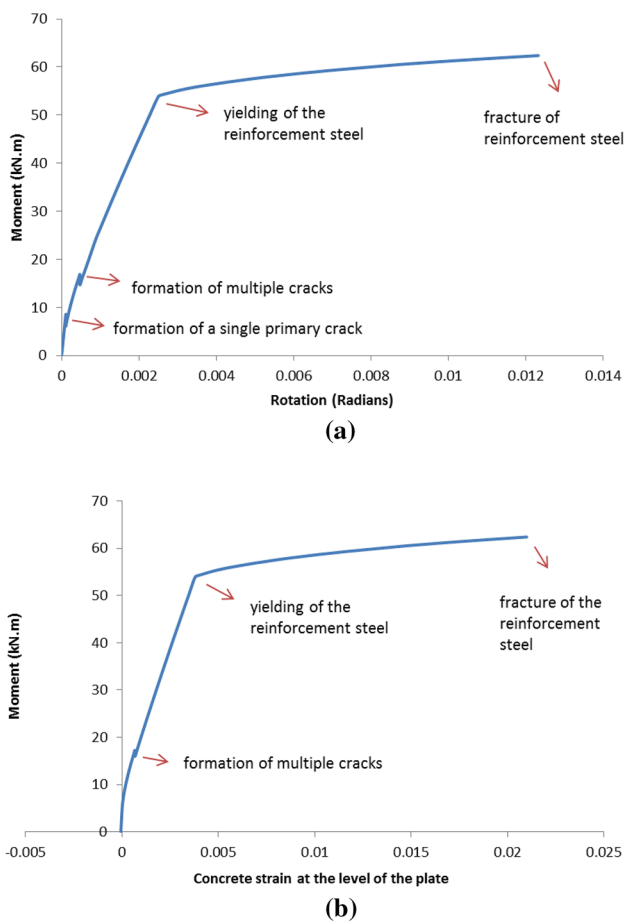


Fig. 18 Passively prestressed segment: (a) M-; (b) M- ϵ_{RC}

the bondstress-slip models shown in Fig. 21a, b were used, respectively. Finally, the load-strain response named “CFRP-B3—Strain Gauge 1” and “CFRP-B3—Strain Gauge 2” represent the experimental results determined in the laboratory for CFRP-B3. From the results, it can be deduced that introducing a frictional component in the bondstress-slip relationship in fact permitted the reinforcement steel to yield at a much higher load in the theoretical solution, and the larger the frictional component the higher the load required to cause the steel reinforcement to yield. As for the strains in the CFRP plate depicted in Fig. 22b, it can be seen that the modified theoretical strains 1 and 2 provide a better solution to the strain in the plate when compared to the theoretical strain response. For instance, the modified theoretical strains tend to approach the recorded experimental strains in the plate and give good correlation when the reinforcement steel yields after a load of about 65 kN. This is opposed the theoretical strain in Fig. 22b where no frictional component is considered in the analysis and the

strain in the plate remains at the IC debonding strain, ϵ_{IC} , after reaching the maximum slip, δ_{max} . Figure 22b illustrates that at service loads, the results correlate poorly with experimental values, indicating that a linear-descending branch does little to simulate the behavior of the plated member at service loads.

Finally, the displacement-based approach was utilized to predict the ultimate load to cause the concrete displacement to equal the plate’s displacement at a distance L_{crit} from the plate’s free end. The concrete displacement can be calculated using Eq. (33) while the displacement of the plate can be determined by;

$$\delta_{db-p} = \int_0^{L_{db}} \epsilon_p \tag{36}$$

where the strain in the plate, ϵ_p , in Eq. (36) can be determined from the segmental analysis. That is, for each moment M , the corresponding ϵ_p can be determined and substituted in Eq. (36). The modified ultimate load obtained when the bondstress-slip model was similar to that in Fig. 21a was 80 kN, and 87.5 kN when the bondstress-slip model was similar to that in Fig. 21b (this while accounting for the self-weight of the beams). The displacement-based ultimate load results are illustrated in Figs. 23 and 24 where the displacement in the FRP plate equaled the displacement in the concrete at the same level at distance L_{crit} from the end of the plate. Clearly, adding a frictional component to the bondstress-slip model allowed the theoretical ultimate load to increase beyond that of the unplated beam and much closer to the average experimental ultimate load (83.9 kN).

6.2 Ascending branch of the bondstress-slip relationship

The poor correlation between the plate’s theoretical strain and experimental strains at low loads (Fig. 22b) can be attributed to the fact that a linear-descending bondstress-slip model was utilized in the analysis. Therefore, to see the effect of adding an ascending branch to the bondstress-slip property on the strain response, a parametric study was conducted.

Initially, the properties of the bondstress-slip model adopted in the analysis above was kept the same (i.e. maximum bondstress, τ_{max} and maximum slip, δ_{max}) with the addition of an ascending branch as depicted in Fig. 25a. The values for the slip at which the maximum bondstress is achieved, δ_1 , were taken as 0.03 mm, 0.05 mm and

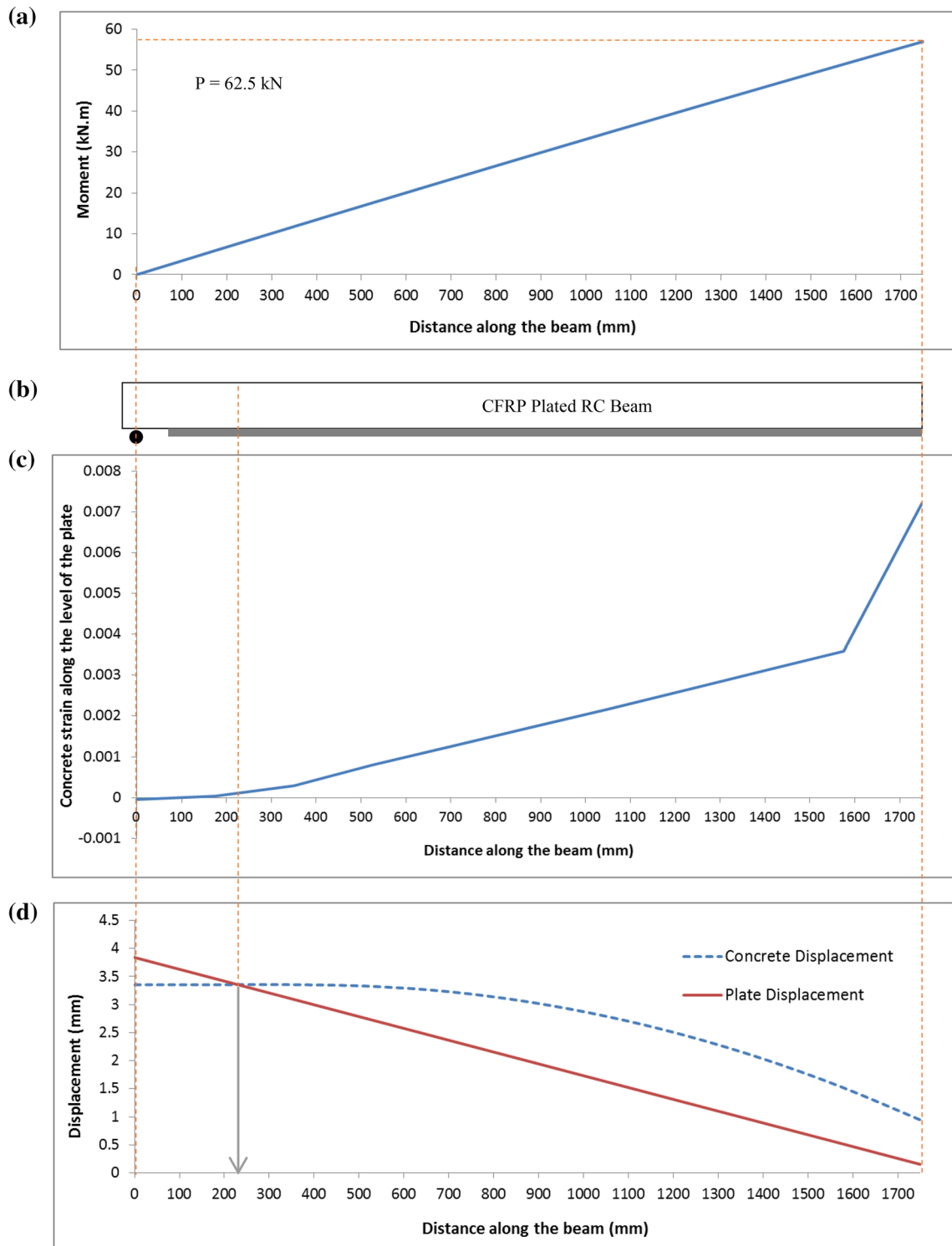


Fig. 19 Displacement-based analysis: **(a)** Moment; **(b)** CFRP plated beam; **(c)** concrete strain along the level of the plate; **(d)** concrete and plate displacements

Table 3 CFRP RC beams experimental and theoretical ultimate loads

Beam specimen	Experimental ultimate load, P_{u-exp} (kN)	Theoretical ultimate load, P_{u-theo} (kN)	P_{u-exp}/P_{u-theo}
CFRP-B1	84.3	62.5	1.35
CFRP-B2	82.8	62.5	1.32
CFRP-B3	84.6	62.5	1.35
Average ultimate load, P_{avg}	83.9	62.5	1.34

Fig. 20 Failure of FRP plated beams [22]

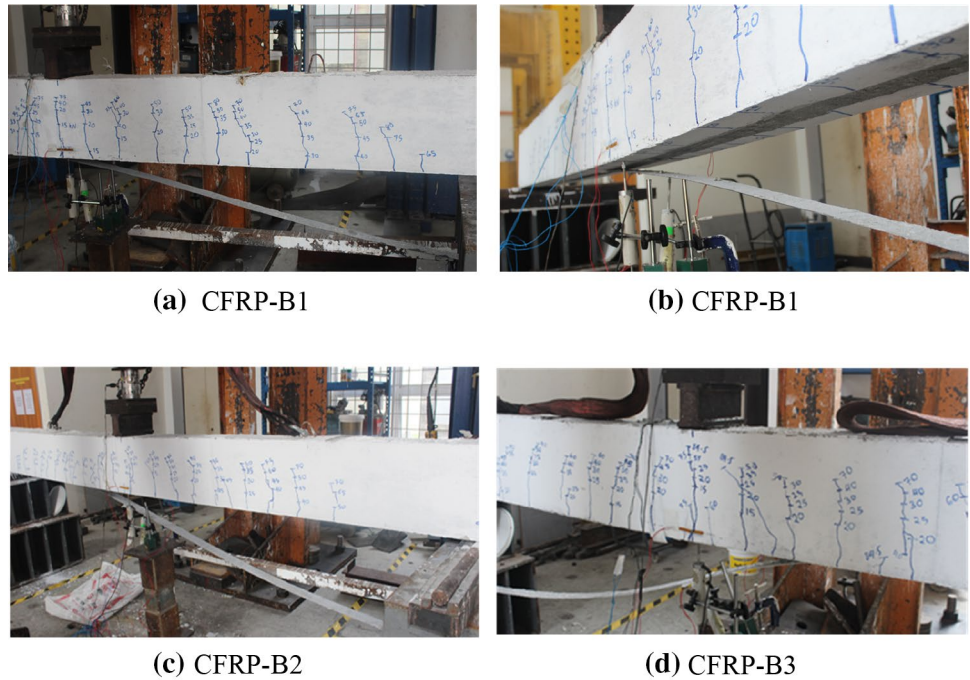
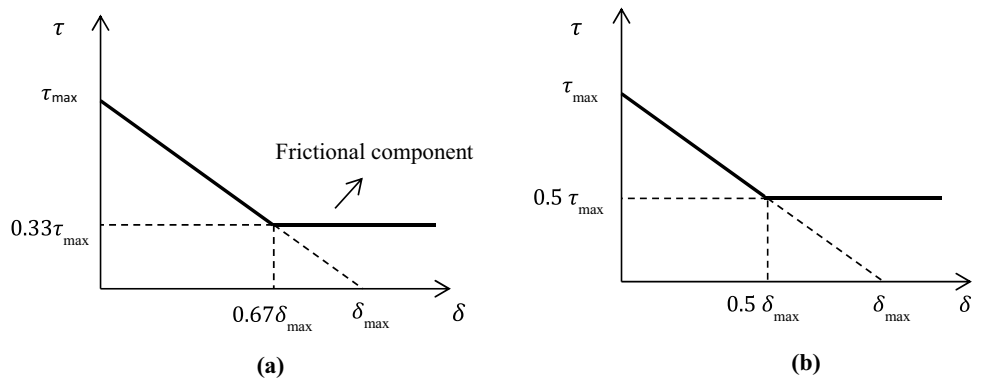


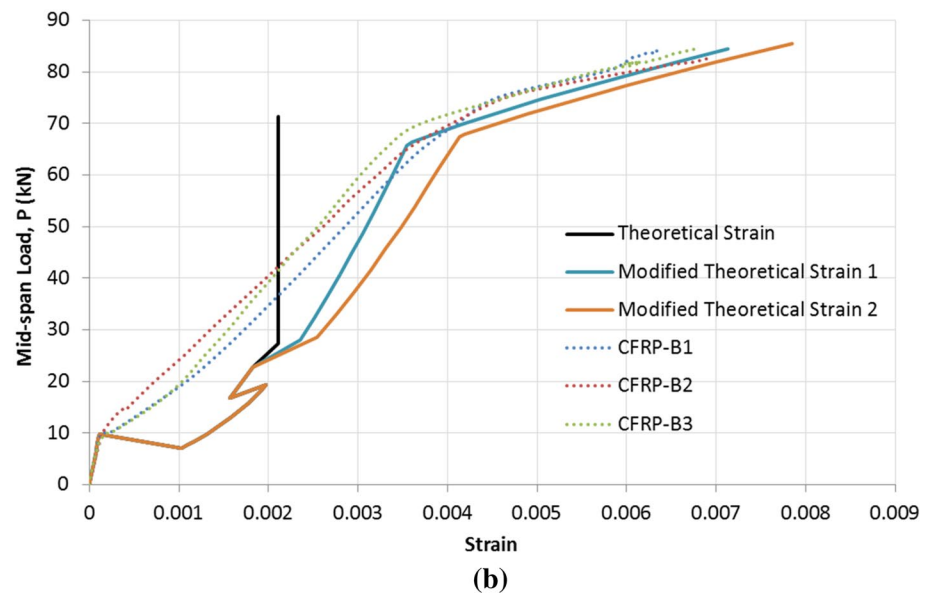
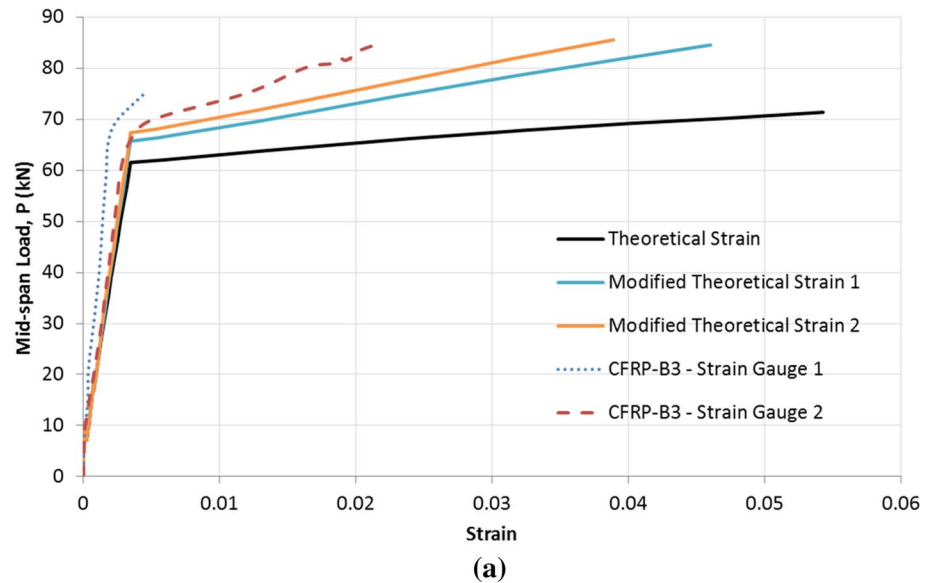
Fig. 21 Proposed bondstress-slip models for CFRP plate



0.07 mm and their effect on the plate strain is illustrated in Fig. 25b. Clearly, the higher the value of δ_1 , the more the theoretical results move towards the experimental results. That is, a higher δ_1 in Fig. 25 meant lower strain development in the plate at service loads. This can be explained by

the fact that as cracks form in the CFRP plated beam, the ascending branch with greater δ_1 develops less bondstress for the same slip; hence, lower force in the plate and lower strain for the same slip (i.e. same applied load). Moreover, adding an ascending branch gives better results at service

Fig. 22 Modified strain results: (a) CFRP-B3 reinforcement steel; (b) CFRP plate



loads than the results derived from the linear-descending branch as can be seen from Fig. 25b. Nevertheless, the change in the plate strain due to different values of δ_1 was not significant as can be seen clearly from the Fig. 25b.

Next, to see the effect of altering the maximum slip on the plate's strain, a parametric study was conducted where the maximum slip, δ_{max} , was changed once to 0.1 mm and then to 0.2 mm and 0.25 mm as shown in Fig. 26a. From the results in Fig. 26b, it can be seen that altering the maximum slip, δ_{max} , does not have much effect on the strain during early stages of loading and the results were

almost identical in all cases. This can be due to the fact that the service behavior of the plated member is heavily influenced by the bondstress at low slip, which was almost the same in all cases as shown in Fig. 26. However, at slightly higher loads, the strain response due to the different bondstress-slip models began to divert, indicating that δ_{max} affects the behavior at higher loads. Furthermore, Fig. 27 shows the effect of changing the maximum bondstress, τ_{max} , on the strain in the plate. Clearly, the lower the τ_{max} , the closer the theoretical results are to the experimental ones as lower strains form in the plate with

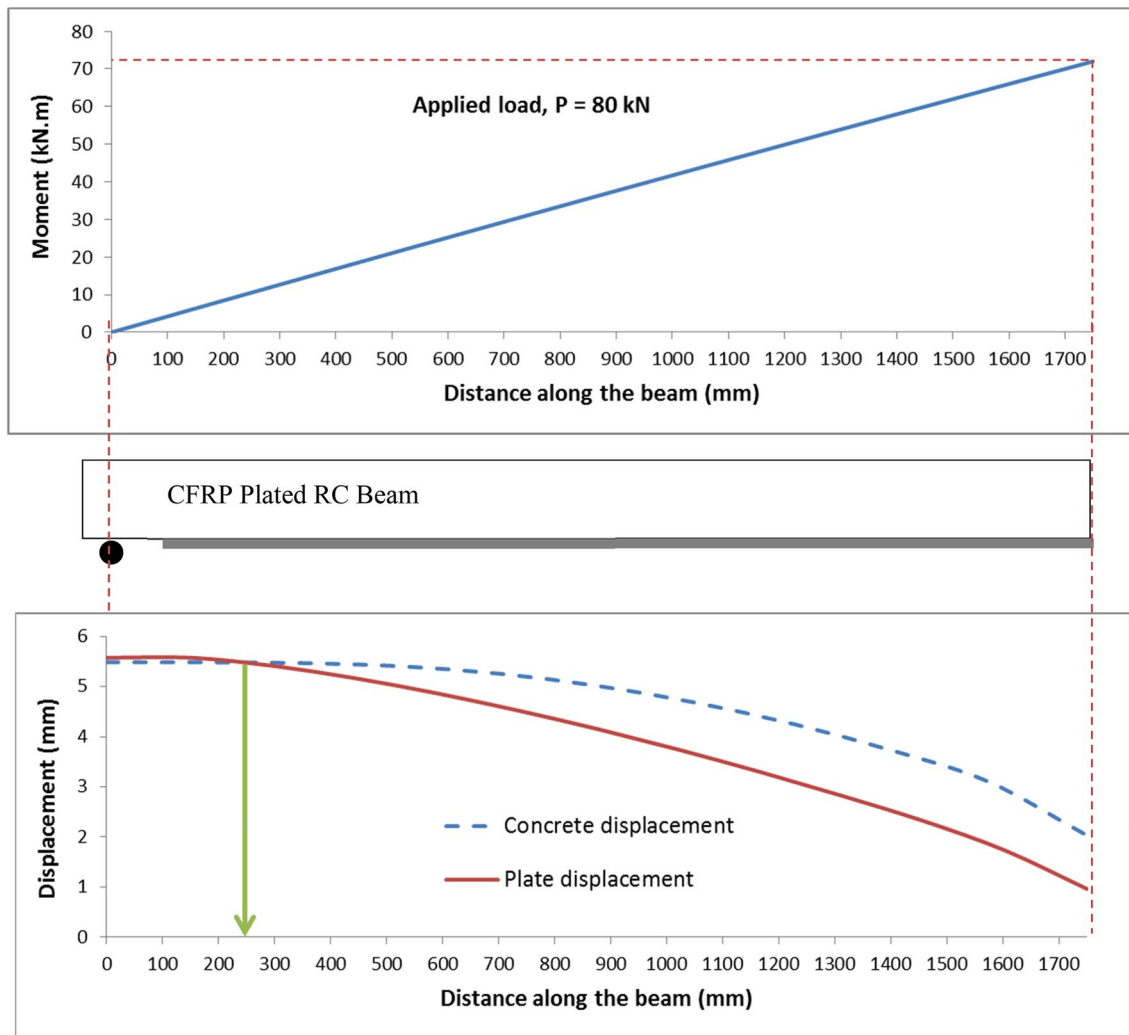


Fig. 23 Ultimate load of CFRP plated RC Beam with 0.33_{max} frictional component

increase in slip (Fig. 27b). Also, altering τ_{max} in the bondstress-slip model significantly influences the behavior at service loads, where the strain response in Fig. 27b varies greatly from one bondstress-slip model to the other.

From the above parametric study, a more accurate bondstress-slip characteristic for the plate that would give better results of the strain at service loads for the beams considered here would have an ascending branch with a considerable δ_1 , and τ_{max} smaller than the 5.23 N/mm^2 used in the preceding analysis. Table 4 gives the values of τ_{max} , δ_1 and δ_{max} based on different bi-linear bondstress-slip models available in the literature, and summarized in Table 5, for the CFRP plated beams considered in this study. The formulas in Table 5 give for each bondstress-slip model the equation for the ascending branch, descending

branch, maximum bondstress, τ_{max} , slip at maximum bondstress, δ_1 , maximum slip, δ_{max} , and the expression for any related variables used. As such, the values in Table 4 were the outcome of direct substitution in the equations given in Table 5. Looking at such results, and bearing in mind the findings of the parametric study, the model presented by Ko et al. [35] gives the greatest δ_1 and lowest τ_{max} among all the models considered in Table 4. Moreover, both Monti et al. [36] and Brosens and Van Gemert [37] models give the greatest value of δ_{max} at approximately 0.44 mm , and Nakaba et al. [38] bondstress-slip model gives the greatest bondstress, τ_{max} , of about 6.33 N/mm^2 . The strain in the CFRP plate against the applied mid-span load, while adopting Ko et al. [35] bondstress-slip model for the plate, is given in Fig. 28.

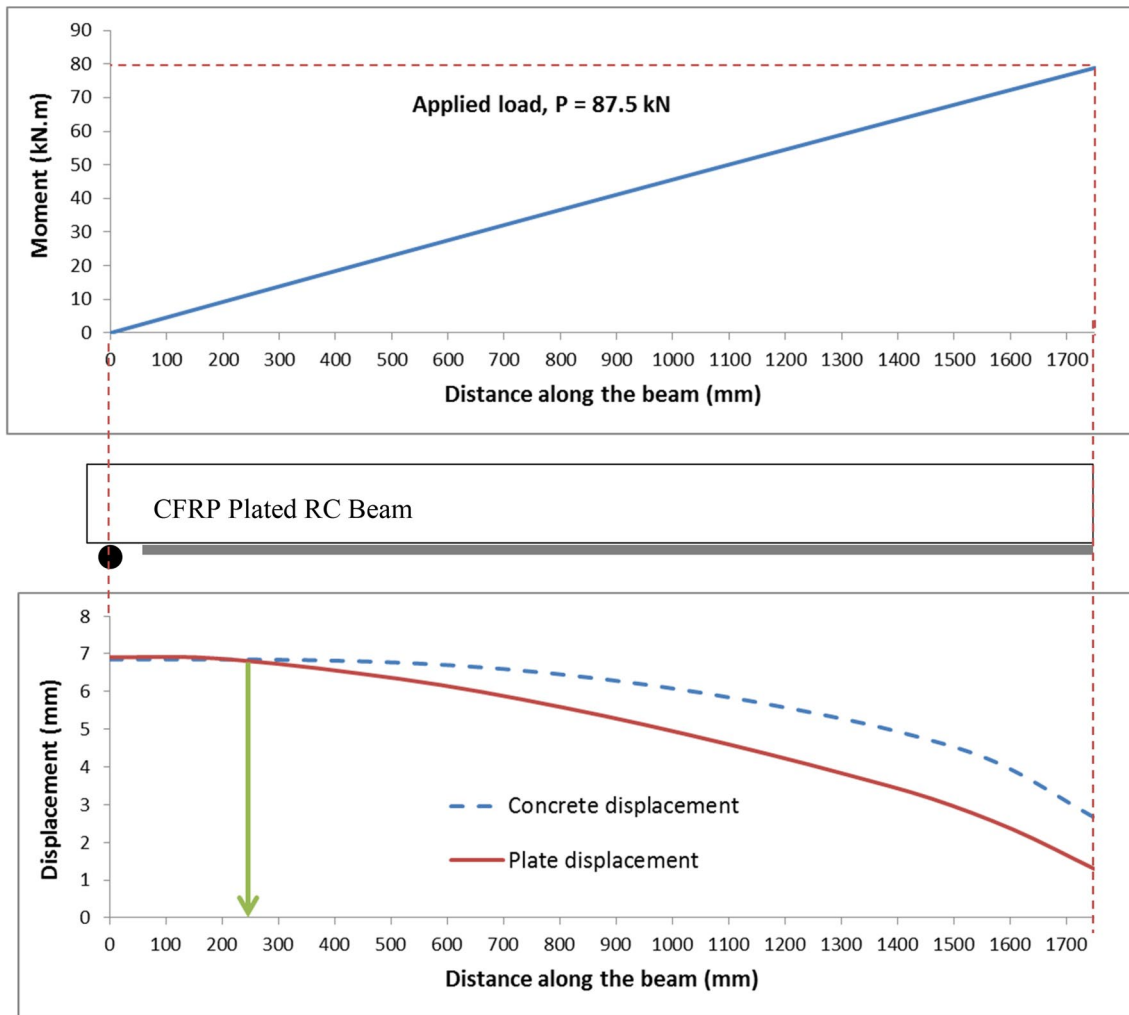


Fig. 24 Ultimate load of CFRP plated RC Beam with 0.5_{max} frictional component

7 Conclusion

Using a displacement-based approach that incorporates member debonding mechanism in the design of CFRP plated beams allows for greater ductility to be achieved in the plate as opposed to a full-interaction approach. In this paper, the analytical results of a displacement-based approach introduced by Oehlers et al. [11] were compared against the experimental values for three CFRP plated RC beam specimens. From the analysis, and due to using a bondstress-slip model for the plate without a residual

shear component, the steel reinforcement yielded at an earlier applied load to that observed in the laboratory. Throughout this paper, it was shown how changing the adopted bondstress-slip model of the plate affects the analytical results and gives better correlation with the experimental findings. The findings of this study can be best summarized as follows:

- Using a linear-descending bondstress-slip model for the FRP plates resulted in poor correlation with the experimental findings at the ultimate limit state. For

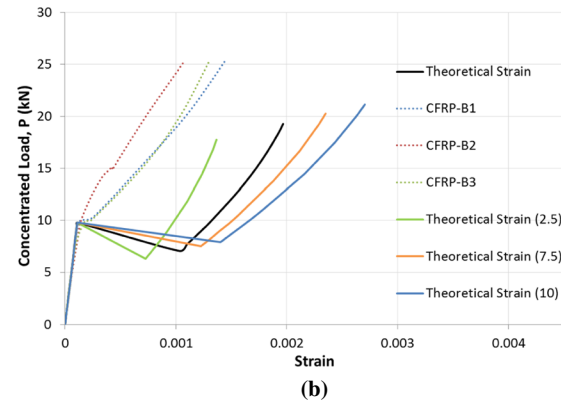
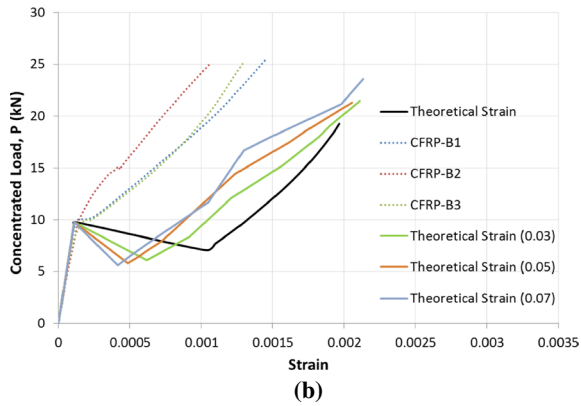
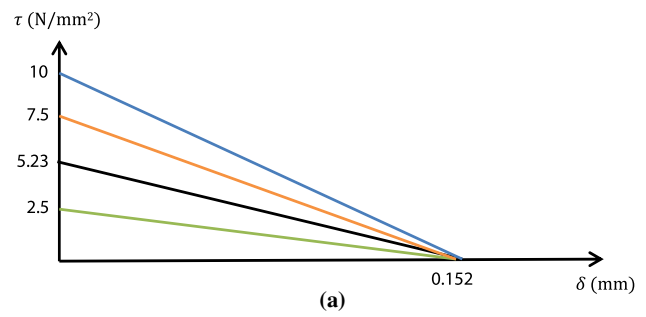
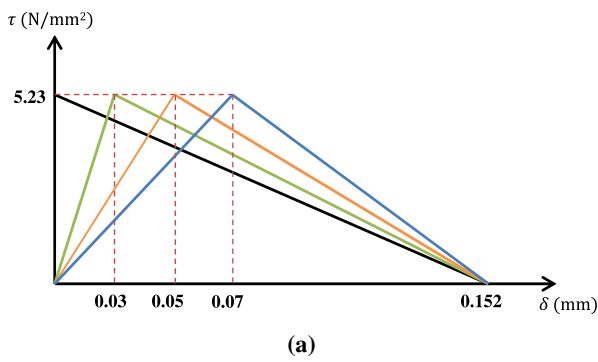


Fig. 25 Altering the slip at maximum bondstress, δ_1

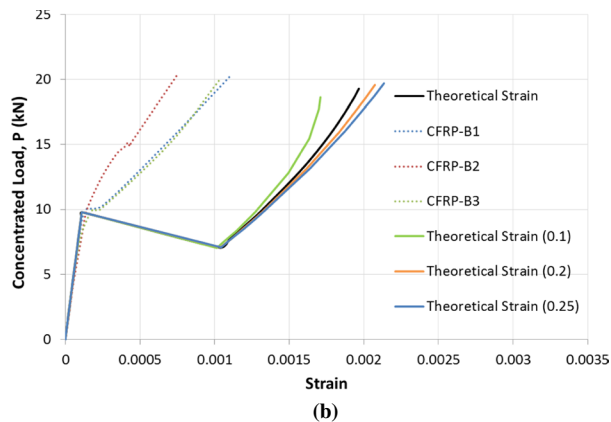
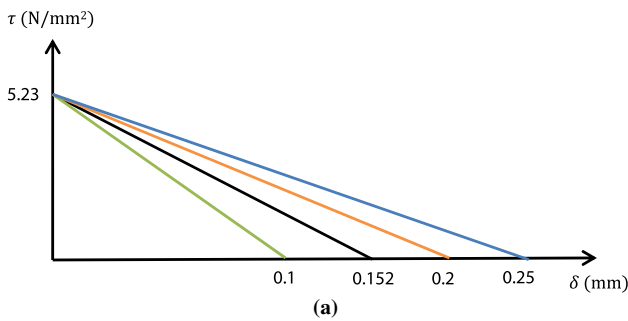


Fig. 26 Altering the maximum slip, δ_{max}

Fig. 27 Altering the maximum bondstress, τ_{max}

Table 4 CFRP plate Bondstress-slip parameters from various models

Ref.	δ_1 (mm)	δ_{max} (mm)	τ_{max} (N/mm ²)
Ko et al. [35]	0.0993	0.257	3.75
Seracino et al. [31]	–	0.152	5.23
Monti et al. [36]	0.0353	0.437	5.93
Nakaba et al. [38]	0.065	–	6.33
Brosens and Van Gemert [37]	0.0353	0.441	5.93

instance, the experimental ultimate load results were on average 34% greater than the theoretical predictions when no frictional component was incorporated in the analysis.

- The higher the adopted residual shear strength component in the FRP plate’s bondstress-slip model, the higher the flexural strength achieved from a displacement-based approach.
- Yielding of the steel reinforcement in the theoretical solution is delayed when incorporating a frictional component in the bondstress-slip model for the plate.
- Adopting a linear-descending bondstress-slip model for the plate gives poor results at service loads, and better predictions can be achieved using bi-linear models.

Table 5 Existing bond-slip models

Bondstress-slip model	Ko et al. [35]	Seracino et al. [31]	Monti et al. [36]
Ascending branch	$\tau_{max} \left(\frac{\delta}{\delta_1} \right)$	–	$\tau_{max} \left(\frac{\delta}{\delta_1} \right)$
Descending branch	$\frac{\tau_{max} (\delta_{max} - \delta)}{(\delta_{max} - \delta_1)}$	$\frac{\tau_{max} (\delta_{max} - \delta)}{(\delta_{max})}$	$\frac{\tau_{max} (\delta_{max} - \delta)}{(\delta_{max} - \delta_1)}$
Maximum bondstress, τ_{max}	$0.165 f_c$	$(0.802 + 0.078 \varphi_f) f_c^{0.6}$	$1.8 \beta_w f_t$
Slip at maximum bondstress, δ_1	$-0.001 f_c + 0.122$	–	$2.5 \tau_{max} \left(\frac{t_a}{E_a} + \frac{50}{E_c} \right)$
Maximum slip, δ_{max}	$-0.002 f_c + 0.302$	$\frac{0.976 \varphi_f^{0.526}}{0.802 + 0.078 \varphi_f}$	$0.33 \beta_w$
Variables	–	$\varphi_f = \frac{d_f}{b_f} = \frac{t_d}{2 t_b + b_p}$ $t_d = t_b = 1$	$\beta_w = \sqrt{\frac{1.5(2 - \frac{b_p}{b_c})}{1 + b_p/100}}$
Type	Bilinear model	Linear-descending model	Bilinear model
Bondstress-slip model	Nakaba et al. [38]	Brosens and Van Gemert [37]	
Ascending branch	$\tau_{max} \left(\frac{\delta}{\delta_1} \right) \left[3 / \left(2 + \left(\frac{\delta}{\delta_1} \right)^3 \right) \right]$	$\tau_{max} \left(\frac{\delta}{\delta_1} \right)$	
Descending branch	–	$\frac{\tau_{max} (\delta_{max} - \delta)}{(\delta_{max} - \delta_1)}$	
Maximum bondstress, τ_{max}	$3.5 f_c^{0.19}$	$1.8 \beta_w f_t$	
Slip at maximum bondstress, δ_1	0.065	$2.5 \tau_{max} \left(\frac{t_a}{E_a} + \frac{50}{E_c} \right)$	
Maximum slip, δ_{max}	–	$\frac{2 G_f}{\tau_{max}}$	
Variables	–	$\beta_w = \sqrt{\frac{1.5(2 - \frac{b_p}{b_c})}{1 + b_p/100}}$, $G_f = 0.3 \beta_w^2 f_t$	
Type	Non-linear model	Bilinear model	

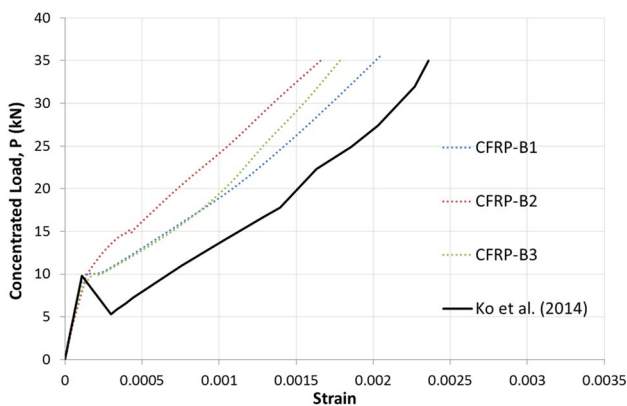


Fig. 28 CFRP plate strain based on Ko et al. [35] bondstress-slip model

- Altering the maximum slip, δ_{max} , in the bondstress-slip model for the FRP plate has a reduced effect on the serviceability response of the plated member.

- Changing the maximum bondstress, τ_{max} , in the bondstress-slip model resulted in a considerable change in the serviceability response of the plated member. For instance, a lower τ_{max} was seen to result in lower service strains while a higher τ_{max} would mean the opposite.
- For the same τ_{max} and δ_{max} , having a greater value of δ_1 in the bondstress-slip model for the FRP plate resulted in lower strains in the plate at low service loads.

Subsequent research on the topic should focus on introducing a bondstress-slip model for EB FRP plates that allows for the force in the plate to keep building up after reaching P_{IC} . This might mean that the bondstress-slip model might have residual shear strength due to friction over the debonded region. Moreover, future research can focus on quantifying the factors that affect the frictional component in bondstress-slip models. Finally, to accurately determine the strains in the plate at service loads, the ascending branch of the bondstress-slip model must be simulated accurately.

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Availability of data and materials Not applicable.

Code availability Not applicable.

Compliance with ethical standards

Conflict of interest The authors declare that they have no conflict of interest.

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