# Development of a new empirical formula for prediction of triple point path 

W. Xiao ${ }^{1}{ }^{(®)} \cdot$ M. Andrae $^{1}{ }^{(®)} \cdot$ N. Gebbeken $^{1}$

Received: 8 November 2019 / Revised: 9 August 2020 / Accepted: 23 August 2020 / Published online: 22 September 2020
© The Author(s) 2020


#### Abstract

This paper develops a new empirical formula for the prediction of the triple point path in irregular shock reflection cases. Numerical simulations using a two-dimensional axisymmetric multi-material arbitrary Lagrangian-Eulerian formulation are employed to obtain the data of fluid density. Using the data of fluid density and nodal coordinates, the gradients of fluid density are determined and then used to generate numerical schlieren images. Based on these images, the triple point paths are derived and compared with the models of the Unified Facilities Criteria (UFC) and Natural Resources Defense Council (NRDC) as well as two models from the open literature. It is found that the numerically derived triple point paths are in good agreement with those predicted by a recently published model in the open literature for the typical ground range of shock wave propagation of up to 6 m . Considering the whole distance range, it is found that the agreement of different models of the triple point path with the numerical ones depends on the considered blast scenario, i.e., the scaled charge height. For small-scaled charge heights, the model of the UFC and the recently published model in the open literature are in better agreement with the numerical results than the other two models, whereas the NRDC model has the best agreement with the numerical results for large-scaled charge heights. Based on the numerical results, a new empirical formula is proposed for the prediction of the triple point path, which is valid for a wide range of the scaled charge heights from 0.5 to $3.5 \mathrm{~m} / \mathrm{kg}^{1 / 3}$ and scaled ground distances up to $15 \mathrm{~m} / \mathrm{kg}^{1 / 3}$.


Keywords Shock wave $\cdot$ Mach stem $\cdot$ Triple point path $\cdot$ Schlieren image $\cdot$ Fluid density

## 1 Introduction

Blast parameters such as peak overpressure and maximum impulse can be calculated from the overpressure-time history. They play a crucial role in the blast-resistant design of structures interacting with shock waves. If a height of burst (HOB) explosion is concerned, the boundary conditions around the shock wave significantly affect the emerging over-pressure-time history (incident blast, ground reflected blast, or Mach stem) and hence influence the protection strategy.

Figure 1a illustrates the blast environment in an HOB explosion. The charge has a mass of $W$ and is detonated

[^0]at a height of $H_{\mathrm{c}}$ above the ground. The angle of incidence $\alpha$ is defined in Fig. 1a, where point A depicts a point of observation on the ground at a distance $R$ from ground zero (GZ). Depending on whether the incident and reflected waves have merged to form a Mach stem or not, the ground range is subdivided into a Mach or a regular reflection region. The triple point $(\mathrm{T})$ is the intersection point of the incident wave (I), reflected wave (R), and Mach stem (M). The origin of the triple point path is denoted as OTP. The quantities $\alpha_{\text {min }}$ and $R_{0}$ are the minimum angle of incidence and the distance for the Mach stem formation, respectively. As the Mach stem travels along the ground surface ( $x$-direction), the triple point rises in the $y$-direction and the height of the triple point $H_{\mathrm{T}}$ increases. More details on the Mach reflection phenomena can be found, for example, in [1-3].

Regarding an HOB explosion, the gauge below the triple point (e.g., point B in Fig. 1a) will experience a single peak (Mach stem) in the overpressure-time history (Fig. 1b), whereas two successive peaks will appear in the overpres-sure-time history (Fig. 1c) measured at the gauge above the


Fig. 1 Height of burst explosions: a blast environment; b overpres-sure-time history at gauge point $\mathrm{B} ; \mathbf{c}$ overpressure-time history at gauge point C
triple point (e.g., point C in Fig. 1a). These two peaks indicate the arrival of the incident and reflected waves, respectively. Thus, significantly different blast loads could be exerted on structures located in different regions. Therefore, an accurate prediction of the triple point path for different blast scenarios is of great importance.

## 2 Models of triple point path

Four models are available in the open literature to calculate the triple point path. They are the Unified Facilities Criteria (UFC) curves (UFC 3-340-02 [4]), the empirical formula of Kinney and Graham [5], Natural Resources Defense Council (NRDC) [6], and Boutillier et al. [7]. UFC published ten curves of the triple point path, for example, Figure 2-13 of UFC 3-340-02 [4]. The scaled charge height varies from 0.4 to $2.8 \mathrm{~m} / \mathrm{kg}^{1 / 3}$. The major drawback of the UFC model is the lack of a formula allowing extrapolation of the triple point prediction for scenarios other than the ten published curves. In addition, the origin of the triple point path is not provided. In order to overcome these two issues, Slavik [8] implemented an engineering model via the function *LOAD_BLAST_ENHANCE in LS-DYNA [9]. Note that all functions used in this study and mentioned in this paper are standard routines of the used software package LS-DYNA.

At first, the UFC curves are used to determine the reflection region (regular or Mach) for HOB explosions. After that, the blast loads are calculated using other charts of UFC 3-340-02 (Figures 2-7, Figures 2-9, and Figures 2-10). The empirical formula of Kinney and Graham [5] is given as
$H_{\mathrm{T}} / H_{\mathrm{c}}=0.07\left(R / R_{0}-1\right)^{2}$,
for $R>R_{0}$. This formula is based on the observations from a variety of experiments using spherical charges. As depicted in Fig. 1a, the distance $R_{0}$ from ground zero to the origin of the triple point path can be calculated as:
$R_{0}=H_{\mathrm{c}} \tan \alpha_{\text {min }}$.

Kinney and Graham [5] proposed an empirical formula to determine the minimum angle of incidence as:
$\alpha_{\text {min }}=1.75 /\left(M_{\mathrm{i}}-1\right)+39$,
where $M_{\mathrm{i}}$ is the Mach number of the incident wave. In order to calculate the distance $R_{0}$ without information of $\alpha_{\min }$, Needham [2] proposed an empirical formula as
$R_{0}=0.825 H_{\mathrm{c}}$,
if $H_{\mathrm{c}} / W^{1 / 3}<99.25 \mathrm{~m} / \mathrm{kt}^{1 / 3}$ and

$$
\begin{align*}
R_{0}= & 170 H_{\mathrm{c}} / W^{1 / 3} /\left[1+25.505\left(H_{\mathrm{c}} / W^{1 / 3}\right)^{0.25}\right] \\
& +1.7176 \cdot 10^{-7}\left(H_{\mathrm{c}} / W^{1 / 3}\right)^{2.5} \tag{5}
\end{align*}
$$

if $H_{c} / W^{1 / 3}>99.25 \mathrm{~m} / \mathrm{kt}^{1 / 3}$, where kt stands for kiloton. The units for the height and the weight in the equation are centimetre and kiloton, respectively. Although this formula is originally obtained from nuclear explosions, it can also be used for conventional high-explosive tests [2, 7]. It should be noted that a high explosive is about twice as effective at forming an air blast compared to a nuclear detonation.

NRDC proposed also a model of the triple point path. It is expressed as:

$$
\begin{equation*}
H_{\mathrm{T}} / W^{1 / 3}=S\left[h+\left(h^{2}+\left(R / W^{1 / 3}-0.9 R_{0}^{\prime}\right)^{2}-R_{0}^{\prime 2} / 100\right)^{0.5}\right] \tag{6}
\end{equation*}
$$

for $R>R_{0}$, where

$$
\begin{align*}
& S=\left[5.98 \cdot 10^{-5}\left(H_{\mathrm{c}} / W^{1 / 3}\right)^{2}+3.8 \cdot 10^{-3} H_{\mathrm{c}} / W^{1 / 3}+0.766\right]^{-1}  \tag{7}\\
& h=0.9 R_{0}^{\prime}-3.6 H_{\mathrm{c}} / W^{1 / 3} \tag{8}
\end{align*}
$$

$R_{0}^{\prime}=\left(H_{\mathrm{c}} / W^{1 / 3}\right)^{2.5} / 5822+2.09\left(H_{\mathrm{c}} / W^{1 / 3}\right)^{0.75}$.
The quantity $R_{0}^{\prime}$ is the scaled quantity of $R_{0}$, i.e.:
$R_{0}^{\prime}=R_{0} / W^{1 / 3}$.

It should be noted that the NRDC model is valid for the scaled charge height from 0 to $8 \mathrm{~m} / \mathrm{kg}^{1 / 3}$ and for charge masses of TNT ranging from 0.1 kt to $25,000 \mathrm{kt}$. The used units for length and mass are metre and kiloton, respectively. Boutillier et al. [7] conducted new experiments in order to develop a new empirical formula to predict the triple point path for smooth and hard reflecting surfaces. It is given as:
$H_{\mathrm{T}} / W^{1 / 3}=a\left(R / W^{1 / 3}\right)^{2}+b R / W^{1 / 3}+c$,
for $R>R_{0}$, where
$a=0.07\left(H_{\mathrm{c}} / W^{1 / 3}\right) / R_{0}^{\prime 2}$,
$b=-2 a R_{0}^{\prime}$,
$c=a R_{0}^{\prime 2}$,
$R_{0}^{\prime}=1.99 \cdot 10^{-3}\left(\mathrm{Eq}_{\mathrm{TNT}}\right)^{1 / 3}\left(H_{\mathrm{c}} / W^{1 / 3}\right)^{2}+0.601 H_{\mathrm{c}} / W^{1 / 3}$.

The quantity $\mathrm{Eq}_{\mathrm{TNT}}$ is the TNT equivalence of C 4 . It is noted that this formula is valid for the scaled charge height from 0.246 to $1.729 \mathrm{~m} / \mathrm{kg}^{1 / 3}$. For this equation to be valid, length and mass are to be given in centimetre and kilogram, respectively.

## 3 Numerical derivation of triple point path

This section aims to develop a new empirical formula to predict the triple point path. Firstly, numerical simulations are conducted with the software LS-DYNA to calculate the fluid density field for different HOB scenarios. Secondly, a corresponding sequence of the numerical schlieren images is generated by using the data of fluid density and nodal coordinates obtained from numerical simulations. These images are employed to obtain the triple point path. Thirdly, a curvefitting method is used to derive the formula to predict the triple point path.

### 3.1 LS-DYNA simulations

Two-dimensional (2D) axisymmetric multi-material arbitrary Lagrangian-Eulerian (MM-ALE) simulations are run


Fig. 2 Two-dimensional axisymmetric MM-ALE model for HOB explosions
with uniformly meshed square elements, comprising the charge and the air domain (Fig. 2). The dimensions of the numerical model are $15 \mathrm{~m} \times 8 \mathrm{~m}$ (length $\times$ height). Free boundary conditions are assumed in the upper and right edges of the numerical model. The nodes at the lower edge of the numerical model are constrained in the normal direction to represent the rigid ground surface. The $y$-axis is the axis of rotational symmetry. A 1 kg TNT charge is centrally detonated at charge height $H_{\mathrm{c}}$ above the ground.

The spherical form of the charge is modelled via the function *INITIAL_VOLUME_FRACTION_GEOMETRY. At first, a one-dimensional (1D) numerical simulation is conducted with an element size of 0.4 mm . This element size provides 131 elements across the charge radius of 1 kg TNT. It is worth pointing out that a sufficiently small element size is vital to accurately model the explosive energy release. As a rule of thumb, at least ten elements across the charge radius should be used for the meshing of the explosive charge [10]. However, at least 50 elements across the charge radius are recommended for modern high-performance computing. In addition, a mesh sensitivity study is performed and three element sizes of $1.6 \mathrm{~mm}, 0.8 \mathrm{~mm}$, and 0.4 mm are used for the study. Figure 3 depicts the time histories of the fluid density at the gauges, which are located at a distance $R$ of 0.75 m and 0.8 m from the charge, respectively. It is observed that the numerical results of the fluid density using all three element sizes are nearly identical. The element size of 0.4 mm is selected for the 1D numerical simulations.

The simulation terminates before the shock wave impinges on the ground surface. Then, the 1D numerical results are mapped to the 2D axisymmetric model (Fig. 2). The use of the mapping techniques is to enable highly fine meshes in the 1D geometry and comparatively coarse meshes in the 2D geometry. They are employed based on two considerations. On the one hand, the accuracy of the numerical results can be ensured. On the other hand, the computational demand is still manageable.

Fig. 3 Fluid density of 1D numerical model, $W=1 \mathrm{~kg}$ TNT: $\mathbf{a} R=0.75 \mathrm{~m} ; \mathbf{b} R=0.8 \mathrm{~m}$


In order to evaluate the effect of the advection method on the numerical results, predictive simulations are conducted with all four advection methods, i.e., the control parameter METH is set to $1,2,-2$, and 3 in the function *CONTROL_ALE. Based on the observations on the density histories at several gauges, $\mathrm{METH}=-2$ is adopted for the 1D numerical simulations. This second-order accurate Van Leer with half index shift (HIS) advection method is recommended by the software LS-DYNA since the function *MAT_HIGH_EXPLOSIVE_BURN is used in the 1D model. The default method ( $\mathrm{METH}=2$ ) is selected for the 2D numerical simulations because all advection methods achieve nearly identical results. Furthermore, the function *MAT_NULL is used instead for the 2D numerical simulations owing to the mapping technique from 1 D to 2 D . Compared to $\mathrm{METH}=2$, the monotonicity condition in METH $=-2$ is relaxed during the advection process, which aims to better preserve the material interfaces for materials defined by *MAT_HIGH_EXPLOSIVE_BURN [9].

The high explosive charge is modelled via *MAT_HIGH_EXPLOSIVE_BURN with the Jones-Wilkins-Lee (JWL) equation of state (EoS). The pertinent material parameters of TNT are listed in Table 1 for the strength model and EoS , where $\rho_{\mathrm{e}}$ is the density of the explosive material, $D$ is the velocity of detonation, $p_{\mathrm{CJ}}$ is the Chapman-Jouguet pressure, $A, B, R_{1}, R_{2}$, and $\omega$ are material constants, and $e_{0}$ is the specific internal energy. The air is modelled via *MAT_NULL and regarded as an ideal gas with a linear polynomial EoS.

In order to verify the 2D axisymmetric numerical model, a mesh sensitivity study is conducted, in which three element sizes of $4 \mathrm{~mm}, 8 \mathrm{~mm}$, and 16 mm are used. Figure 4 displays the time histories of the fluid density at the gauges, which are located at a horizontal distance $R$ of $2 \mathrm{~m}, 4 \mathrm{~m}, 8 \mathrm{~m}$, and 12 m from the charge, respectively. To better indicate the
possible mesh differences, Fig. 5 is created for the integral of the overdensity, i.e., fluid density minus the initial density $\left(1.225 \mathrm{~kg} / \mathrm{m}^{3}\right)$. The gauge height $h$ is 0.5 m above the ground. The HOB scenario of $W=1 \mathrm{~kg}$ TNT and $H_{\mathrm{c}}=1 \mathrm{~m}$ is used here. It is observed that the numerical results (both the fluid density and the integral of the overdensity) using all three element sizes are nearly identical. The element size of 4 mm is selected for further analyses. Thus, the numerical model has in total 7.5 million elements. The termination time of 2D axisymmetric numerical simulations is 40 ms . Standard atmospheric conditions are assumed in the numerical analyses, i.e., at mean sea level and at the standard temperature of $15^{\circ} \mathrm{C}$. This results in an atmospheric pressure of 101.325 kPa and an air density of $1.225 \mathrm{~kg} / \mathrm{m}^{3}$ [12].

### 3.2 Numerical schlieren images

Based on the data of fluid density ( $\rho$ ) and nodal coordinates $(x, y)$, a corresponding sequence of numerical schlieren images is generated. These data are obtained from numerical simulations at predefined tracer points in the ALE mesh. Since the tracer points are defined as fixed in space (via TRACK $=1$ in the function *DATABASE_TRACER), the tracer point coordinates are identical to the nodal coordinates. The distance between the tracer points is 2 cm . Hence, in total 0.3 million tracer points are defined. They are used to determine the gradient of fluid density, which can be used to define the shock profiles. The magnitude $|\nabla \rho|$ of the gradient of fluid density is calculated as:
$|\nabla \rho|=\sqrt{(\partial \rho / \partial x)^{2}+(\partial \rho / \partial y)^{2}}$.
Furthermore, a nonlinear shading function $\Phi$ is used to accentuate the weak flow features,
$\Phi=\exp \left(-k|\nabla \rho| /|\nabla \rho|_{\max }\right)$,

Table 1 Material parameters of the strength model and EoS for TNT [11]

| $\rho_{\mathrm{e}}\left(\mathrm{kg} / \mathrm{m}^{3}\right)$ | $D(\mathrm{~m} / \mathrm{s})$ | $p_{\mathrm{CJ}}(\mathrm{Pa})$ | $A(\mathrm{~Pa})$ | $B(\mathrm{~Pa})$ | $R_{1}(-)$ | $R_{2}(-)$ | $\omega(-)$ | $e_{0}(\mathrm{~Pa})$ |
| :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- |
| 1630 | 6930 | $2.10 \mathrm{E}+10$ | $3.712 \mathrm{E}+$ <br> 11 | $3.231 \mathrm{E}+$ <br> 09 | 4.15 | 0.95 | 0.30 | $7.00 \mathrm{E}+09$ |


where $|\nabla \rho|_{\text {max }}$ is the maximum value of the gradient of fluid density and $k$ is a constant used for the flow visualization, in which the greyscale fringe levels in the numerical schlieren images are accurately adjusted. The darker the image, the larger the gradient of fluid density. More description of the numerical schlieren algorithm can be found in [13].

Figure 6 demonstrates how the triple point can be optically detected by means of numerical schlieren images for the scenario of $W=1 \mathrm{~kg}$ TNT and $H_{\mathrm{c}}=1 \mathrm{~m}$. In this case, a value of $k=20$ is used for the flow visualization. At $t=0.5 \mathrm{~ms}$ (Fig. 6a), the incident wave (I) propagates spherically. Once the shock wave impinges on the ground surface, a reflected wave ( R ) from the ground surface is generated. At a distance of 1.0 m from ground zero, the Mach stem begins to form, i.e., at $t=1.1 \mathrm{~ms}$ (Fig. 6b). It is indicated by the intersection of the incident (I) and reflected (R) waves at the point OTP on the ground surface (see zoomed view in Fig. 6b). As the shock
wave further propagates, the triple point (T) rises (Fig. 6c-f). Below the triple point (T), the Mach stem (M) is developed by the coalescence of the incident $(\mathrm{I})$ and reflected $(\mathrm{R})$ waves. In addition, not only the primary incident $\left(\mathrm{I}_{1}\right)$ and reflected $\left(R_{1}\right)$ waves but also the secondary incident $\left(I_{2}\right)$ and reflected $\left(R_{2}\right)$ waves are evident in the flow field (Fig. 6d-f). Applying this method to consecutive numerical schlieren images, the triple point path is derived. This method implies that a Mach stem only becomes visible once the selected distance between the tracer points is smaller than or equal to 2 cm .

Using the concept of comparison in [7], the numerically derived triple point paths are illustrated in Fig. 7 (labelled as "Schlieren"), compared to the other four models of triple point path (labelled as "Kinney and Graham", UFC, NRDC, and "Boutillier et al.", respectively). Six different blast scenarios are involved, in which the charge mass ( $W=1 \mathrm{~kg} \mathrm{TNT}$ ) is detonated at a height $H_{\mathrm{c}}$ from

Fig. 5 Integral of the overdensity of 2D axisymmetric numerical model, $W=1 \mathrm{~kg}$ TNT and $H_{\mathrm{c}}=1 \mathrm{~m}$, gauge height $h=0.5 \mathrm{~m}$ : a $R=2 \mathrm{~m}$; b $R=4 \mathrm{~m} ; \mathbf{c} R=8 \mathrm{~m}$; d $R=12 \mathrm{~m}$

0.5 to 3.0 m above the ground surface. Using (1)-(3), the triple point paths labelled as "Kinney and Graham" are determined. The triple point paths predicted by the UFC model are obtained from numerical simulations using the function *LOAD_BLAST_ENHANCE. Note that the reference curves for the UFC data presented in this paper were taken from graphs and not determined by an equation. For $H_{\mathrm{c}}=1.5-2.5 \mathrm{~m}$ (Fig. 7c-e), the UFC curves are truncated since no available data exist for large ground distances. Furthermore, Fig. 7f has no triple point path predicted by the UFC model since the scaled charge height ( $3.0 \mathrm{~m} / \mathrm{kg}^{1 / 3}$ ) is outside of the applicability range ( $2.8 \mathrm{~m} / \mathrm{kg}^{1 / 3}$ ) of the UFC model. It should be noted that the NRDC model is applicable for charge masses ranging from 0.1 kt to $25,000 \mathrm{kt}$ TNT, which are much larger than the charge mass ( 1 kg ) assumed in this study. This fact is most likely attributed to the disagreement in the triple point results between the NRDC model and the other models.

For the typical ground range of shock wave propagation of up to 6 m , the numerical triple point paths are in good accordance with those predicted by the model of Boutillier et al. [7]. This is because the properties of the reflecting sur-
face assumed in this model (smooth and hard) and in the numerical simulations (smooth and rigid) are nearly identical. However, the difference in the height of the triple point predicted by the model of Boutillier et al. and by numerical schlieren images increases, as the distance from ground zero becomes large. For $H_{c}=2.0-3.0 \mathrm{~m}$ (Fig. 7d-f), the triple point paths predicted by the model of Boutillier et al. are illustrated as dashed-dotted lines, since the associated scaled charge heights are actually larger than their maximum applicability range ( $1.729 \mathrm{~m} / \mathrm{kg}^{1 / 3}$ ).

Considering the whole distance range, the numerical triple point path for $H_{\mathrm{c}}=0.5 \mathrm{~m}$ (Fig. 7a) agrees well with the ones predicted by the models of UFC and Boutillier et al. For $H_{\mathrm{c}}=$ $1.5-3.0 \mathrm{~m}$ (Fig. 7c-f), it agrees reasonably well with the ones predicted by the NRDC model. For $H_{\mathrm{c}}=1.0 \mathrm{~m}$ (Fig. 7b), it lies between the ones predicted by the Boutillier et al. and NRDC models. It is noted that, for this charge height, the triple point paths predicted by the models of Boutillier et al. and Kinney and Graham are identical. In general, the NRDC model predicts a triple point path serving as the lower bound, whereas the UFC (Kinney and Graham) model serves as the upper bound for small (large)-scaled charge heights. The dif-

Fig. 6 Detection of triple points using numerical schlieren images, $W=1 \mathrm{~kg}$ TNT and $H_{\mathrm{c}}=1 \mathrm{~m}: \mathbf{a} t=0.5 \mathrm{~ms}$; b $t=1.1 \mathrm{~ms} ; \mathbf{c} t=3.5 \mathrm{~ms}$; $\mathbf{d} t=6.5 \mathrm{~ms} ; \mathbf{e} t=9.1 \mathrm{~ms}$; f $t=13.9 \mathrm{~ms}$

ference between these models of the triple point path is most likely due to the different properties (roughness and hardness) of the ground surface at the test sites of the respective experiments used for the development of the models [7]. It is worth pointing out that the models of Kinney and Graham, UFC, and NRDC had not provided any information on the roughness and hardness of the reflecting surface [7].

However, the location of the triple point path depends not only on the charge height and mass but also on the roughness and hardness of the reflecting surface. It was revealed in extensive experiments (e.g., in [14-16]) that high surface roughness delayed the Mach stem formation and lowered the height of the triple point. It is noted that as $H_{\mathrm{c}}$ becomes larger (e.g., 3.0 m , Fig. 7f), the numerical triple point path is likely

Fig. 7 Comparison of different models of triple point paths, $W=1 \mathrm{~kg}$ TNT: $\mathbf{a} H_{\mathrm{c}}=0.5 \mathrm{~m}$; b $H_{\mathrm{c}}=1.0 \mathrm{~m} ; \mathbf{c} H_{\mathrm{c}}=1.5 \mathrm{~m}$; d $H_{\mathrm{c}}=2.0 \mathrm{~m} ; \mathbf{e} H_{\mathrm{c}}=2.5 \mathrm{~m}$; f $H_{\mathrm{c}}=3.0 \mathrm{~m}$

to fall below the ones predicted by the models of Kinney and Graham and Boutillier et al. This is most likely related to the small scale of the numerical calculations and the difficulty of resolving the shock fronts at low overpressure and overdensity as well as their gradients.

In summary, the observations mentioned above reveal that the agreement of the models of the triple point path with the numerical ones depends on the considered blast scenario, i.e., the scaled charge height. For small-scaled charge heights, the models of UFC and Boutillier et al. are in better agreement with the numerical results than the other two models (Kinney and Graham, NRDC), whereas the NRDC model has the best agreement with the numerical results for large-scaled charge heights. These differences confirm the intention of this study, i.e., that a new tool for the prediction of the triple point path is required, which is applicable to a wide range of ground distances and charge heights.

### 3.3 Curve-fitting method

The triple point path is likely to follow a parabolic-like shape. If the charge height increases or the charge mass decreases, the triple point path flattens. Furthermore, as observed from the UFC curves, the scaled height of the triple point decays exponentially with the scaled charge height. The model of Kinney and Graham depicts the dependence of the height of the triple point on the charge height and the distance from ground zero. The formulae of NRDC and Boutillier et al. describe this dependence as the scaled height of the triple point versus the scaled charge height and the scaled ground distance. Unlike the formulae of NRDC and Boutillier et al., the UFC model provides the scaled height of the triple point versus the scaled ground distance as individual curves for different scaled charge heights. The intent of this study is to propose a new empirical formula, which aims to incorporate
the dependence of the scaled height $H_{\mathrm{T}} / W^{1 / 3}$ of the triple point both on the scaled charge height $H_{c} / W^{1 / 3}$ and on the scaled distance $R / W^{1 / 3}$ from ground zero into a single equation. Furthermore, it is applicable to a wide range of ground distances and charge heights. For general use, this formula is given based on the scaled quantities $\left(H_{\mathrm{T}} / W^{1 / 3}, H_{\mathrm{c}} / W^{1 / 3}\right.$, and $R / W^{1 / 3}$ ) as

$$
\begin{align*}
& \frac{H_{\mathrm{T}}}{W^{1 / 3}}=\left(\mathrm{e}^{-d \frac{H_{\mathrm{c}}}{W^{1 / 3}}}+f\right)\left\{\frac{R}{W^{1 / 3}}-\left[a\left(\frac{H_{\mathrm{c}}}{W^{1 / 3}}\right)^{2}+b \frac{H_{\mathrm{c}}}{W^{1 / 3}}+c\right]\right\}_{18)}^{2} \\
& \quad \text { for } R>R_{0}, \tag{18}
\end{align*}
$$

where
$R_{0}=W^{1 / 3}\left[a\left(\frac{H_{\mathrm{c}}}{W^{1 / 3}}\right)^{2}+b \frac{H_{\mathrm{c}}}{W^{1 / 3}}+c\right]$.
Based on the numerical data for blast scenarios, i.e., charge mass $W=1 \mathrm{~kg}$ TNT and charge height $H_{\mathrm{c}}=0.5-3.5 \mathrm{~m}$, the coefficients $a, b, c, d$, and $f$ of (18) are determined using the MATLAB Curve Fitting Toolbox ${ }^{\mathrm{TM}}$ [17], i.e., $a=0.2104$, $b=0.6003, c=0.248, d=2.534$, and $f=0.009393$. The coefficient of determination is the square of the correlation between the response values and predicted response values. In this case, it is equal to 0.9998 , which indicates that there is a good correlation between the numerically derived values of the heights of the triple point and the ones predicted by (18). It should be noted that only the data of triple points before the shock wave impinges on the boundaries are used for the development of the empirical formula.

Figure 8 illustrates the fitted surface versus the numerical data used for the calibration. The scaled charge height $H_{\mathrm{c}} / W^{1 / 3}$ varies from 0.5 to $3.5 \mathrm{~m} / \mathrm{kg}^{1 / 3}$. The scaled distance $R / W^{1 / 3}$ varies from 0 to $15 \mathrm{~m} / \mathrm{kg}^{1 / 3}$. It is worth mentioning that two assumptions are made during the derivation of the new empirical formula. One is the scaled height of the triple point decays exponentially with the scaled charge height, which is based on the observation from the UFC curves. Another assumption is that the scaled height of the triple point has a parabolic relationship with the scaled distance from ground zero, which is proposed by Boutillier et al. [7].

It is worth pointing out that two main differences exist between this new empirical formula and the one of Boutillier et al. Firstly, Boutillier et al. [7] used own experimental data to develop the empirical formula. This study uses data from numerical simulations. This provides more data points for the curve fitting. The formula of Boutillier et al. is valid for the scaled charge height up to $1.729 \mathrm{~m} / \mathrm{kg}^{1 / 3}$ and for scaled ground distances up to $6 \mathrm{~m} / \mathrm{kg}^{1 / 3}$. By virtue of numerical simulations, the application range of (18) is extended to $3.5 \mathrm{~m} / \mathrm{kg}^{1 / 3}$ for the scaled charge height and $15 \mathrm{~m} / \mathrm{kg}^{1 / 3}$ for the scaled ground distance. Secondly, for the formula


Fig. 8 Fitted surface versus numerical data of triple points
of Boutillier et al., there are in total five equations required for the estimation of the triple point path. In addition, if the scaled charge height is changed, the coefficients of the equations of this model, specifically, $a, b, c$, and $R_{0}^{\prime}$, need to be re-calculated. In this study, only one equation is required for the estimation of the triple point path, and the coefficients of the equation do not depend on the scaled charge height. In this sense, this new empirical formula is simpler.

The triple point paths predicted by (18) are also plotted in Fig. 7 (labelled as CFM) for the different blast scenarios. As observed in Fig. 7, the fitted curves are in good agreement with the numerically derived triple point paths. In order to evaluate the reliability of the formula developed in this study, a possibility would be the use of the experimental data, especially for large ground distances. However, such experimental data are scarce in the open literature.

## 4 Summary

This paper proposes a new empirical formula to predict the triple point path in HOB explosion scenarios. Two-dimensional axisymmetric multi-material arbitrary Lagrangian-Eulerian numerical simulations are conducted to calculate the fluid density field. Using the data of fluid density and nodal coordinates, the gradients of fluid density are determined and used to generate the numerical schlieren images. Based on these images, the triple point paths are derived, which are in good agreement with those predicted by the model of Boutillier et al. [7] for the typical ground range (up to 6 m ) of the shock wave propagation used in this model. Considering the whole distance range, it can be seen that the agreement of different models of the triple point path with the numerical ones depends on the considered blast scenario, i.e., the scaled charge height. For small-scaled charge
heights, the models of UFC and Boutillier et al. are in better agreement with the numerical results than the other two models (Kinney and Graham, NRDC), whereas the NRDC model has the best agreement with the numerical results for large-scaled charge heights. Based on the numerical data of triple points, a new empirical formula is proposed to derive the triple point path. It is valid for the range of scaled charge heights from 0.5 to $3.5 \mathrm{~m} / \mathrm{kg}^{1 / 3}$ and scaled distances up to $15 \mathrm{~m} / \mathrm{kg}^{1 / 3}$. For general use, this formula is given in terms of the scaled quantities. Unlike the available models of the triple point path, the dependencies of the scaled height of the triple point both on the scaled charge height and on the scaled distance from ground zero are incorporated into a single equation.

Acknowledgements The authors would like to thank the Federal Office of Civil Protection and Disaster Assistance (BBK), Department II.5-Structural Protection, Emergency Preparedness (Water) in Germany for their financial support to carry out this research work.The funding was provided by Bundesamt für Bevölkerungsschutz und Katastrophenhilfe (Grant No. BA2533).

Funding Open Access funding provided by Projekt DEAL.

Open Access This article is licensed under a Creative Commons Attribution 4.0 International License, which permits use, sharing, adaptation, distribution and reproduction in any medium or format, as long as you give appropriate credit to the original author(s) and the source, provide a link to the Creative Commons licence, and indicate if changes were made. The images or other third party material in this article are included in the article's Creative Commons licence, unless indicated otherwise in a credit line to the material. If material is not included in the article's Creative Commons licence and your intended use is not permitted by statutory regulation or exceeds the permitted use, you will need to obtain permission directly from the copyright holder. To view a copy of this licence, visit http://creativecomm ons.org/licenses/by/4.0/.

## References

1. Ben-Dor, G.: Shock Wave Reflection Phenomena. Springer, Berlin (2007). https://doi.org/10.1007/978-3-540-71382-1
2. Needham, C.E.: Blast Waves, 2nd edn. Springer, Berlin (2018). https://doi.org/10.1007/978-3-319-65382-2
3. Gebbeken, N., Döge, T.: Vom Explosionsszenario zur Bemessungslast. Der Prüfingenieur 29, 42-52 (2006). (in German)
4. US Army Corps of Engineers: Structures to resist the effects of accidental explosions. UFC 3-340-02, 2nd edn. Air Force Civil Engineer Support Agency (2014)
5. Kinney, G.F., Graham, K.J.: Explosive Shocks in Air. Springer, Berlin (1985). https://doi.org/10.1007/978-3-642-86682-1
6. McKinzie, M.G., Cochran, T.B., Norris, R.S., Arkin, W.M.: The US Nuclear War Plan: A Time for Change. Nat. Resour. Def. Counc., New York (2001)
7. Boutillier, J., Ehrhardt, L., de Mezzo, S., Deck, C., Magnan, P., Naz, P., Willinger, R.: Evaluation of the existing triple point path models with new experimental data: proposal of an original empirical formulation. Shock Waves 28(2), 243-252 (2018). https://doi. org/10.1007/s00193-017-0743-7
8. Slavik, T.P.: A coupling of empirical explosive blast loads to ALE air domains in LS-DYNA ${ }^{\circledR}$. IOP Conf. Ser. Mater. Sci. Eng. 10, 012146 (2010). https://doi.org/10.1088/1757-899X/10/1/012146
9. Livermore Software Technology Corp.: LS-DYNA—Keyword User's Manual, Version R10.0 (2017)
10. Schwer, L., Rigby, S.: Secondary and height of burst shock reflections: application of afterburning. In: Proceedings of the 25th Intern. Symp. on Military Aspects of Blast and Shock, Paper No. 003, Zürich, Switzerland, Spiez Laboratory (1983)
11. Dobratz, B.M., Crawford, P.C.: LLNLExplosives Handbook: Properties of Chemical Explosives and Explosive Simulants. Lawrence Livermore National Laboratory, University of California, Oakland (1985)
12. Xiao, W., Andrae, M., Gebbeken, N.: Experimental and numerical investigations of shock wave attenuation effects using protective barriers made of steel posts. J. Struct. Eng. 144(11), 4018204 (2018). https://doi.org/10.1061/(ASCE)ST.1943-541X. 0002194
13. Quirk, J.J., Karni, S.: On the dynamics of a shock-bubble interaction. J. Fluid Mech. 318, 129-163 (1996). https://doi.org/10.1017/ S0022112096007069
14. Bryant, E.J., Eberhard, R.A., Kingery, C.N.: Mach Reflection Over Hard Packed Dirt and Dry Sand. Report No. 809, Ballistic Res. Lab., Aberdeen Proving Ground (1952)
15. Huber, P.W., McFarland, D.R.: Effect of Surface Roughness on Characteristics of Spherical Shock Waves. Technical report, Accession No. ADA395014, Defense Technical Information Center (1955)
16. Reisler, R.E., Pettit, B.A.: Project DIPOLE WEST-Multiburst Environment (Non-simultaneous Detonations). Final report, Accession No. ADA031985 (1976)
17. The Math Works Inc.: MATLAB Curve Fitting Toolbox User's Guide (2014)

Publisher's Note Springer Nature remains neutral with regard to jurisdictional claims in published maps and institutional affiliations.


[^0]:    Communicated by F. Seiler.
    W. Xiao
    weifang.xiao@unibw.de
    1 RISK Research Center, Institute of Engineering Mechanic and Structural Analysis, University of the Bundeswehr Munich, Werner-Heisenberg-Weg 39, 85577 Neubiberg, Germany

